

Figure 1.1: A schematic diagram of a shock tunnel

I represents the initial shock into an equilibrium test gas.

R represents the reflected shock with an equilibrium nozzle reservoir behind it.

S represents the standing shock on a wedge in a nonequilibrium test gas.

e denotes equilibrium conditions

f denotes frozen conditions

r denotes relaxing conditions

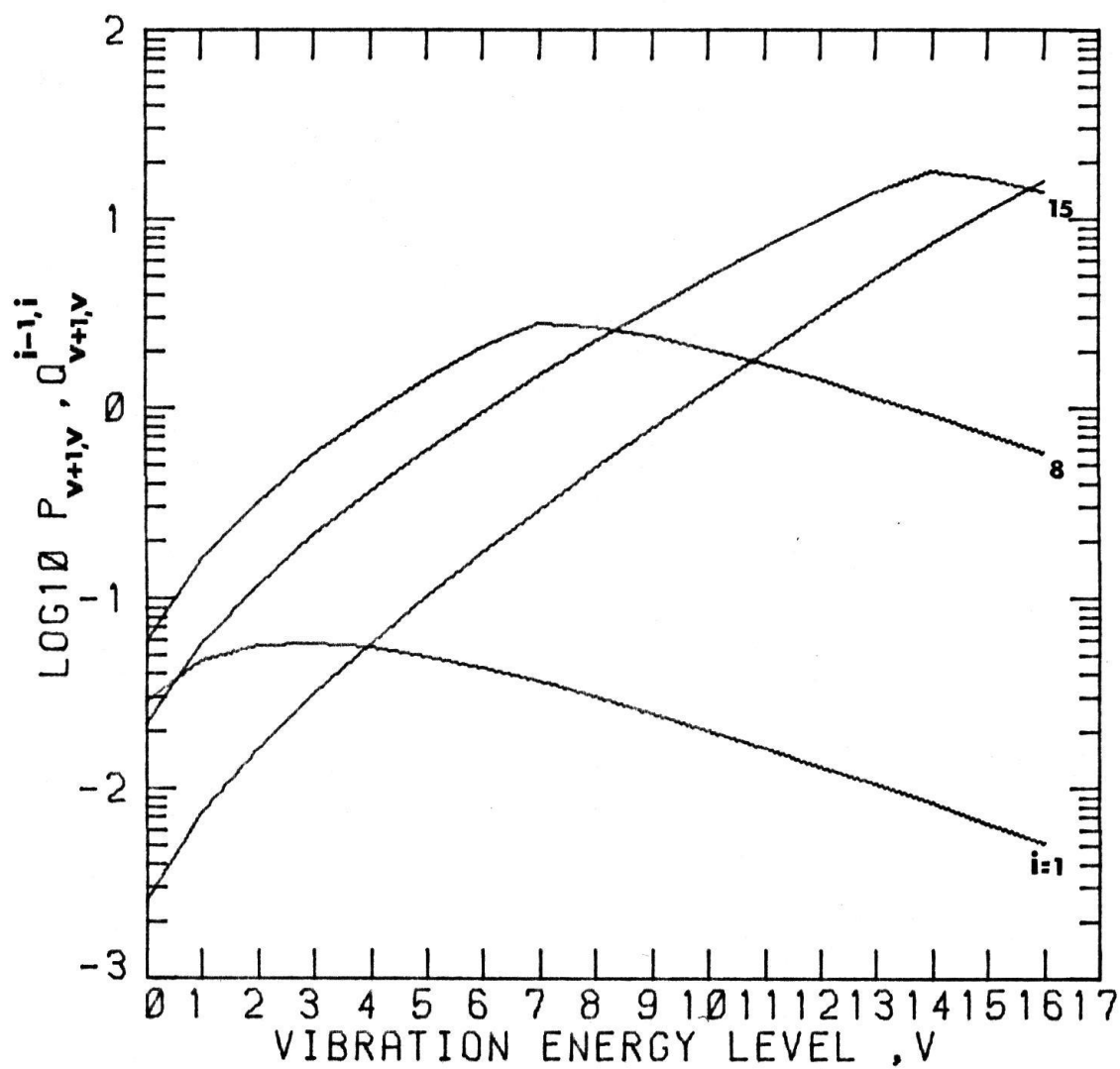


Figure 2.1: Vibrational transition probabilities for H₂ at 5000K and 1 atm.

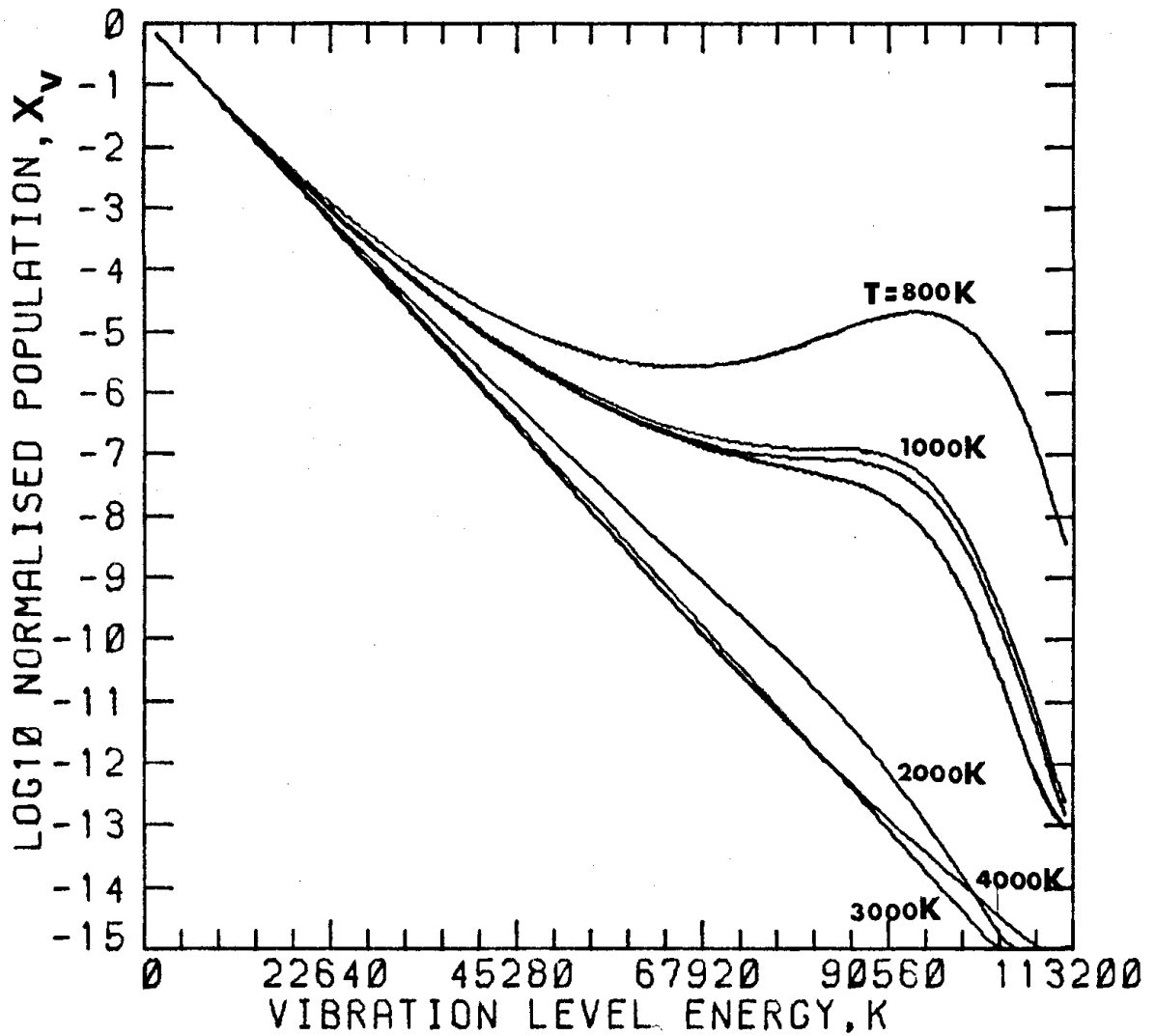


Figure 2.2: Vibrational level population distribution for N_2 at 1 atm and $Y = 3000K$. The translational temperature varies from 4000 to 800K. The two curves just below $T = 1000K$ correspond to values of -3×10^5 and $-3 \times 10^6 Ks^{-1}$ for $\frac{dT}{dt}$, respectively.

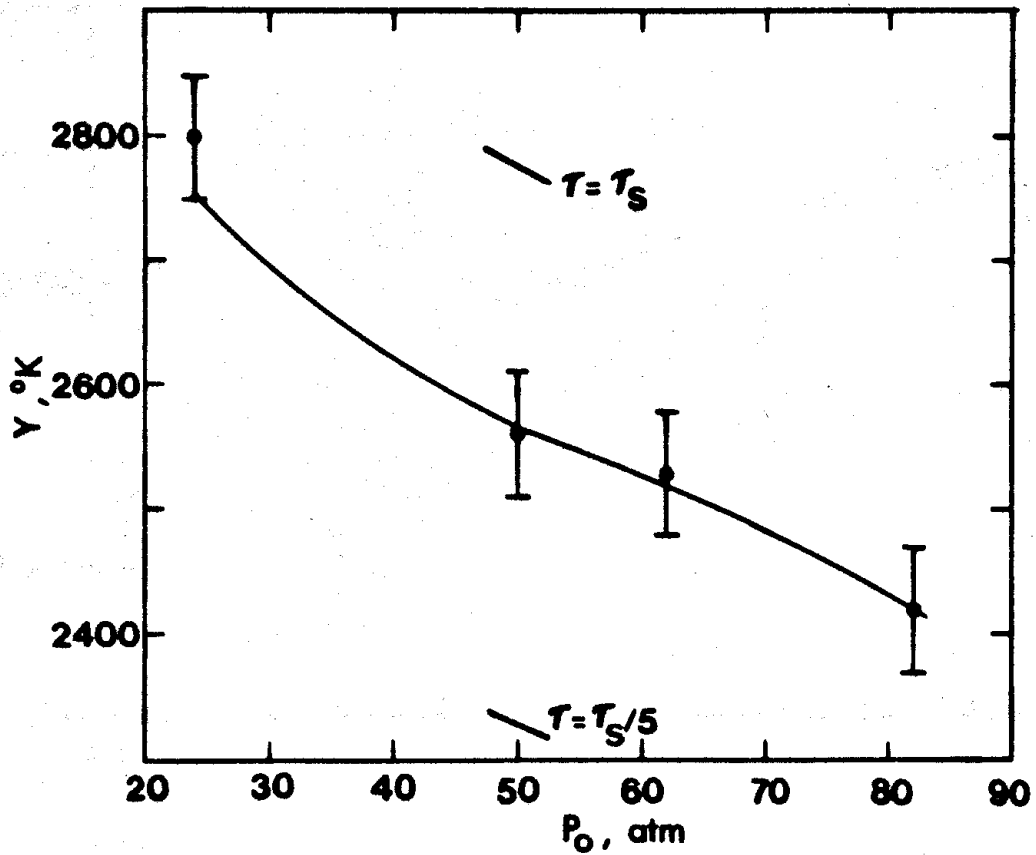


Figure 2.3: Vibrational temperature (Y) vs reservoir pressure (P_0) at a reservoir temperature $T_0 = 4500\text{K}$ and area ratio $A/A^* = 32$. — Theory; ● experimental value of Y from Hurle et al. (1964) reinterpreted according to MacDonald (1972).

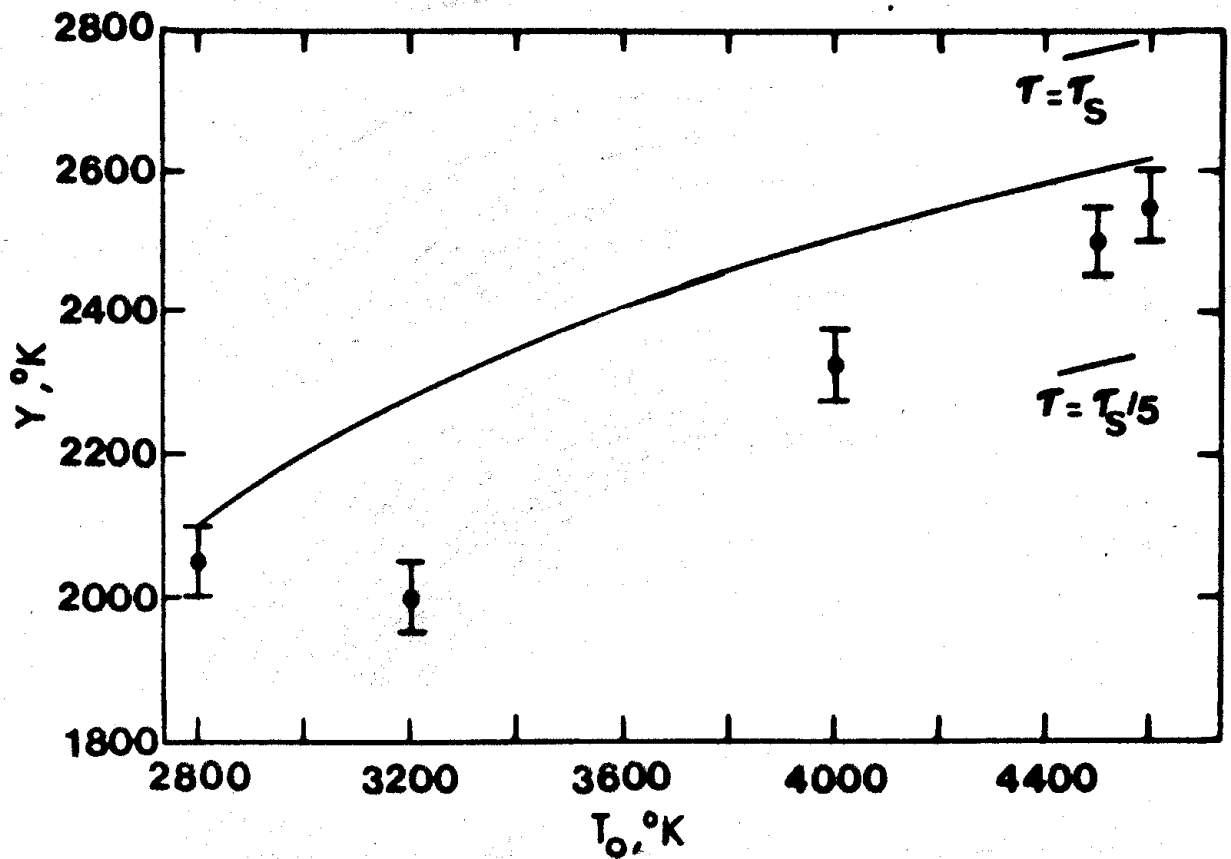


Figure 2.4: Vibrational temperature (Y) vs reservoir temperature (T_0) at $P_0 = 50$ atm and area ratio $A/A^* = 8$. — Theory; ● experimental value of Y from Hurle et al. (1964) reinterpreted according to MacDonald (1972).

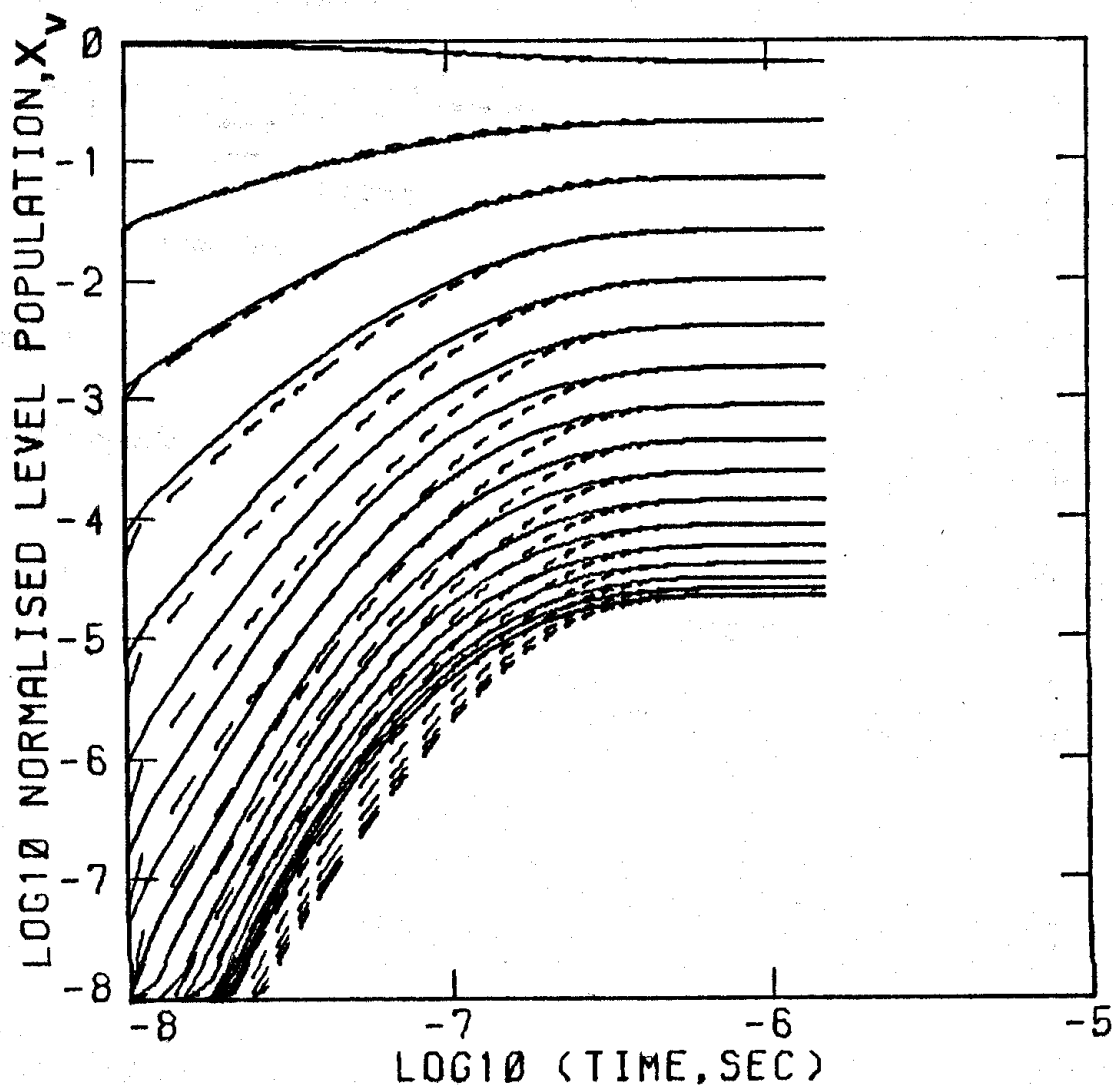


Figure 2.5: Time evolution of vibrational level populations of H_2 at 5000K and 1 atm with no dissociation. — V-T transitions only, --- V-T and V-V transitions. The populations were in a Boltzmann distribution at 300K.

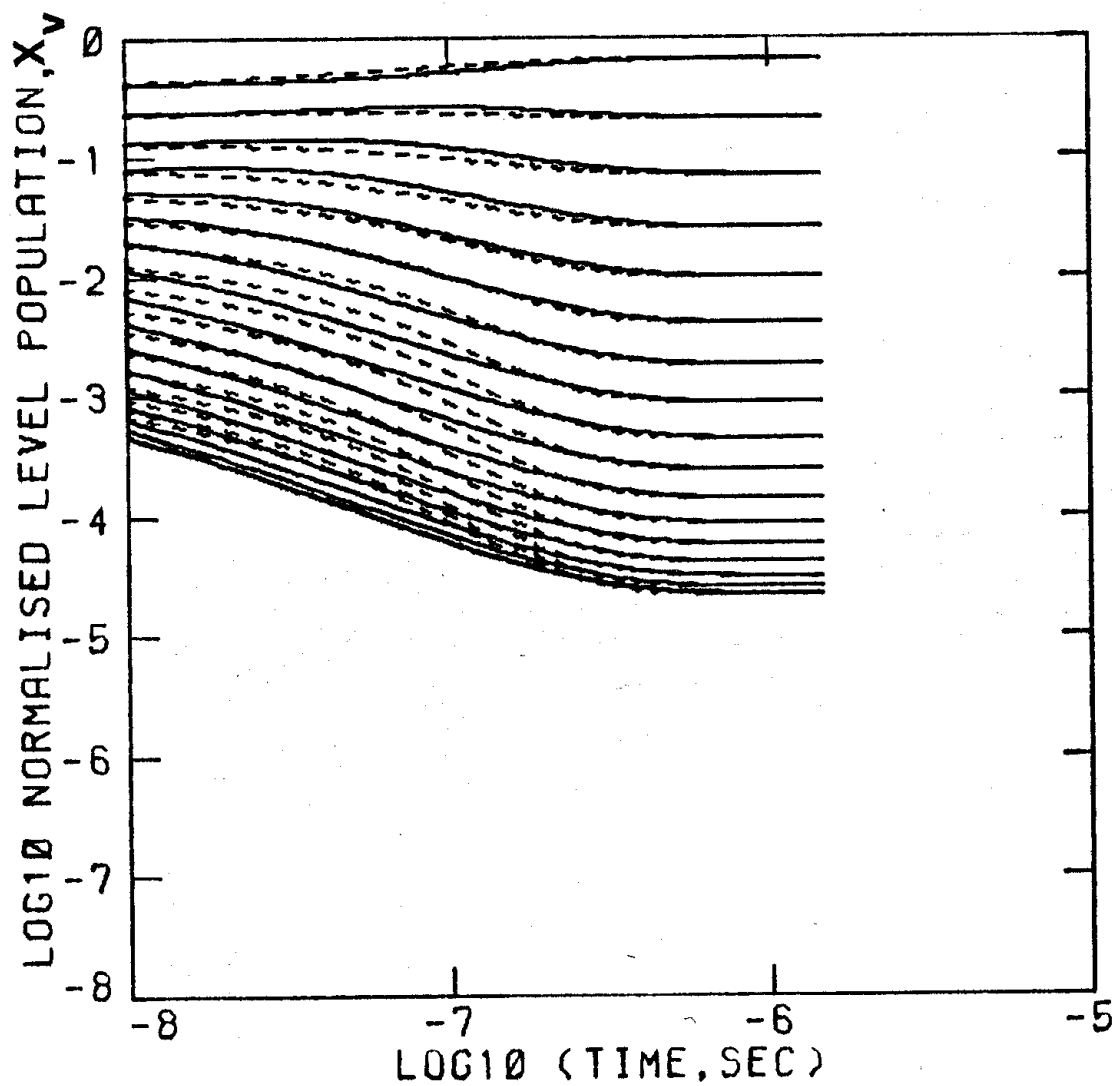


Figure 2.6: Time evolution of vibrational level populations of H₂ at 5000K and 1 atm with no recombination.—V-T transitions only, --- V-T and V-V transitions. The populations were in a Boltzmann distribution at 10000K.

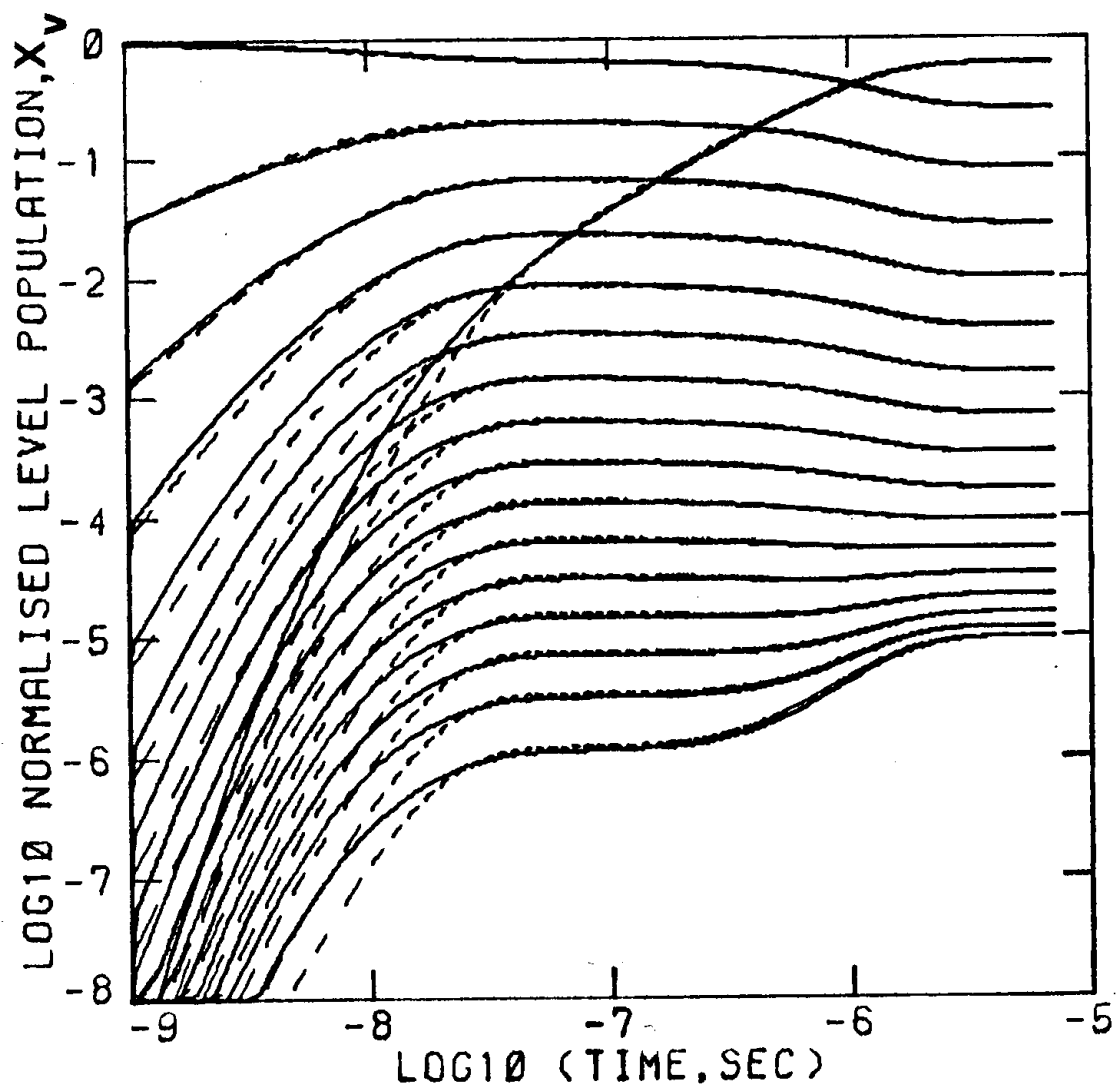


Figure 2.7: Time evolution of vibrational level populations of H_2 at 5000K and 10 atm with dissociation. Half the atom population is also plotted. — V-T transitions only, --- V-T and V-V transitions. The populations were in a Boltzmann distribution at 300K.

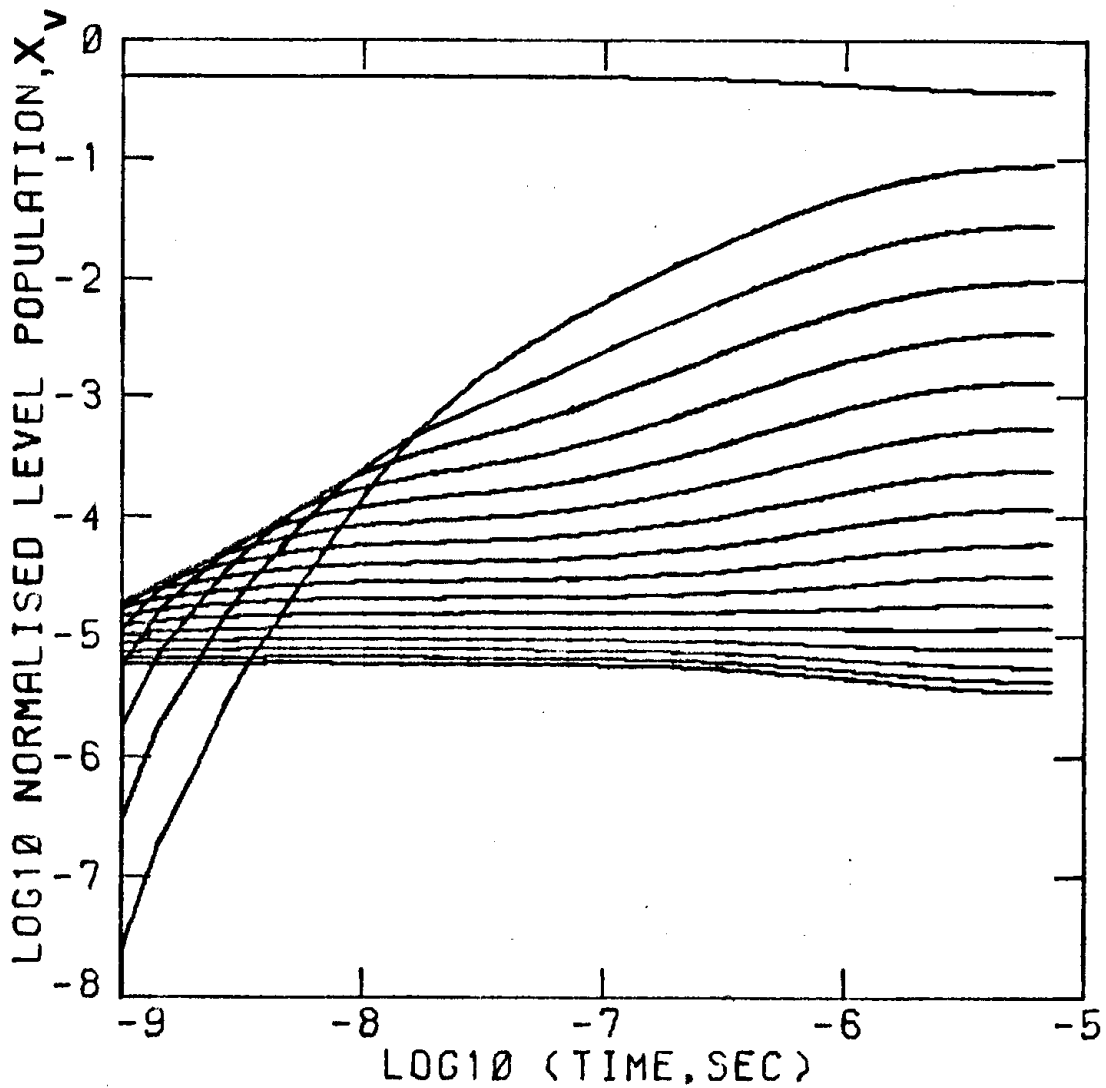


Figure 2.8: Time evolution of vibrational level populations of H_2 at 5000K and 10 atm with recombination. Half the atom population is also plotted. — V-T transitions only, --- V-T and V-V transitions. The populations were in Boltzmann distribution at 10000K. The mole fraction of H_2 was 10^{-8} .

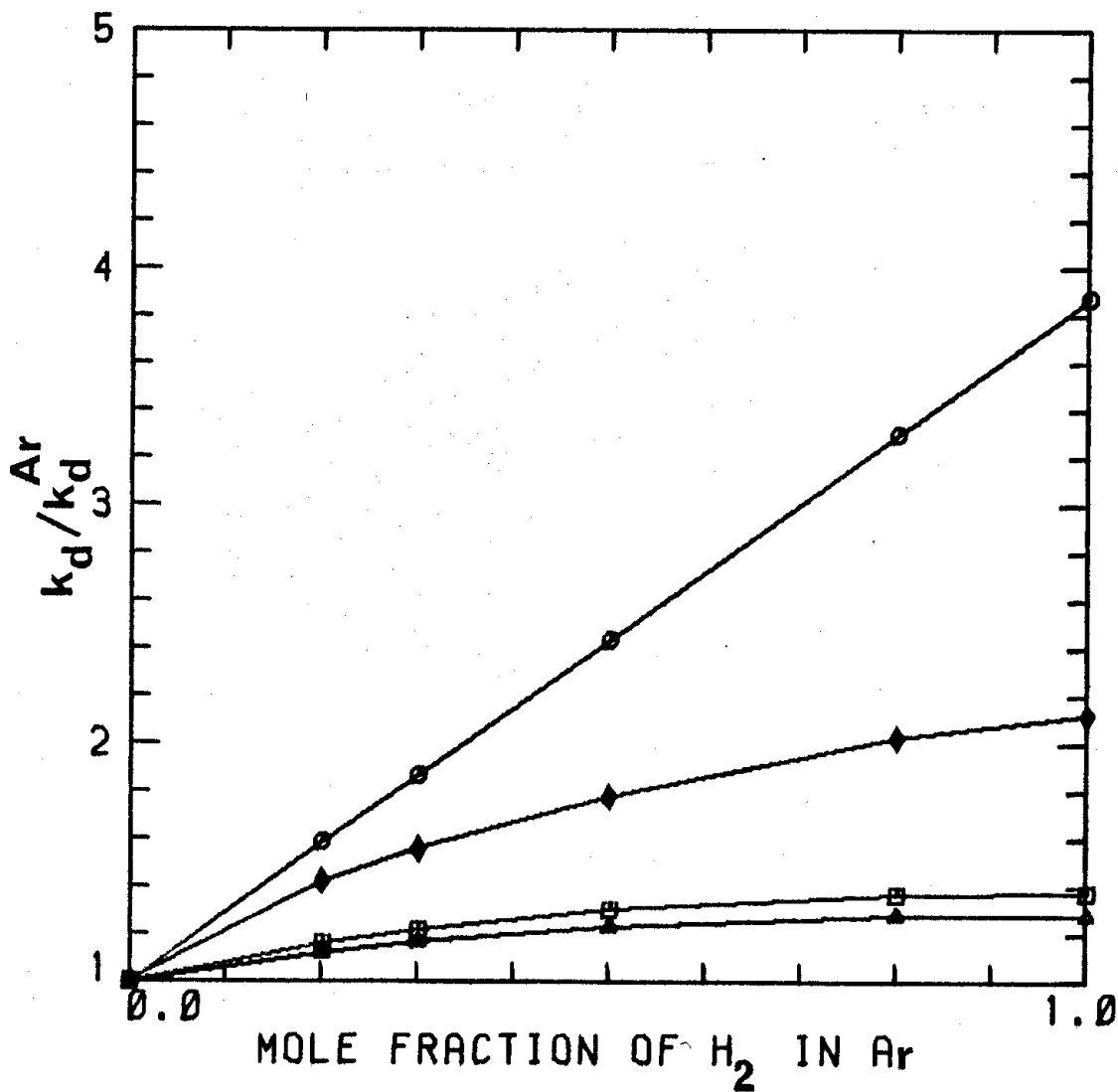


Figure 2.9: Variation of initial dissociation rate of H₂ with mole fraction of H₂ in Ar at 5000K and 1 atm. —▲— V-T transitions only, —◻— V-T and V-V transitions, —◆— V-T and V-V transitions with q equal to 100, —○— V-T transitions only, with equal normalised V-T transition probabilities for H₂-H₂ and H₂-Ar collisions.

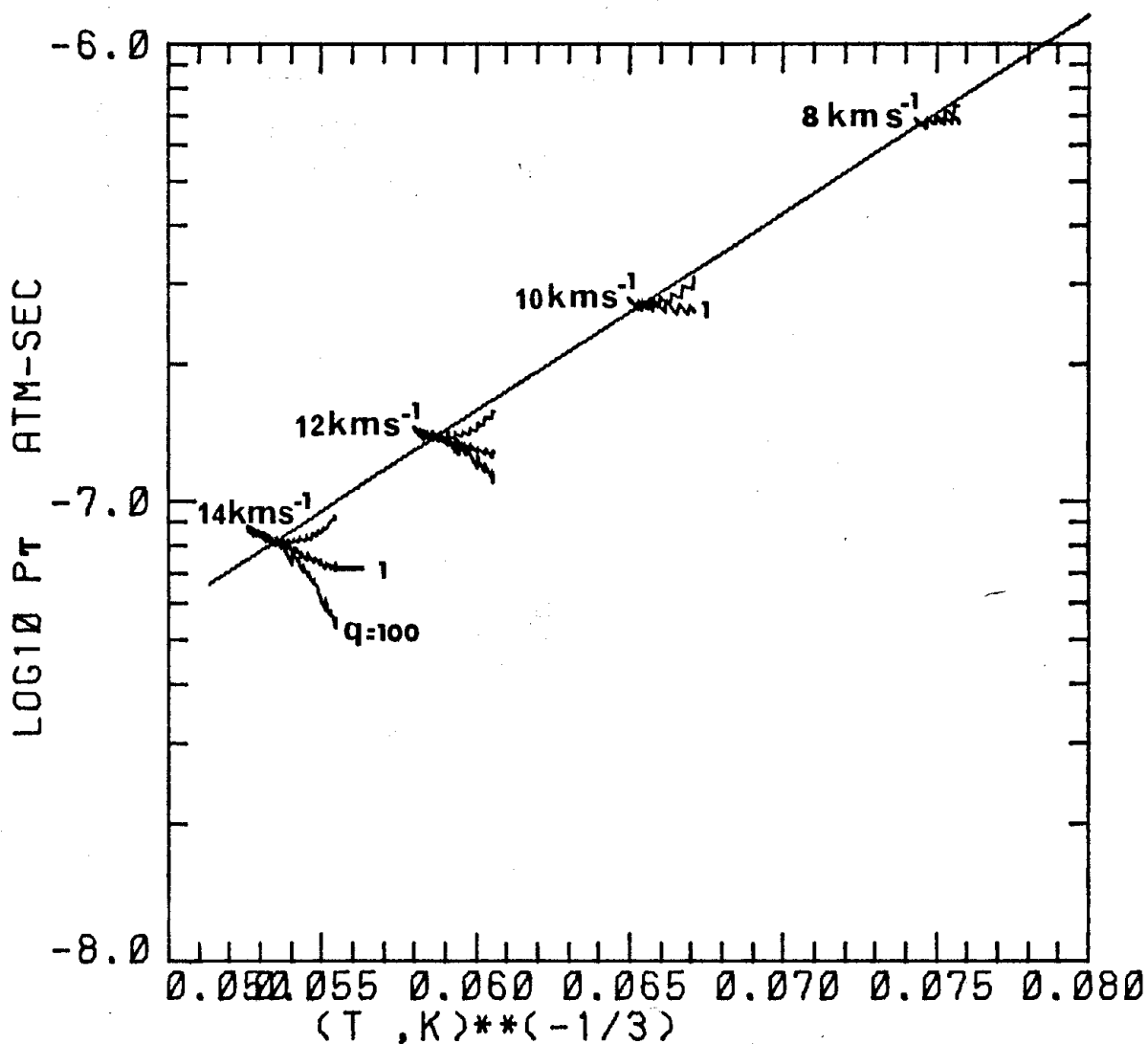


Figure 2.10: Vibrational excitation rates of H_2 . The straight line is the experimental value of Kiefer and Lutz (1966b). Results of four shocks in 5 torr initial pressure of H_2 at 300K are plotted. The slowest rate for each shock is for V-T transitions only. The effect of V-V transitions is shown by varying q .

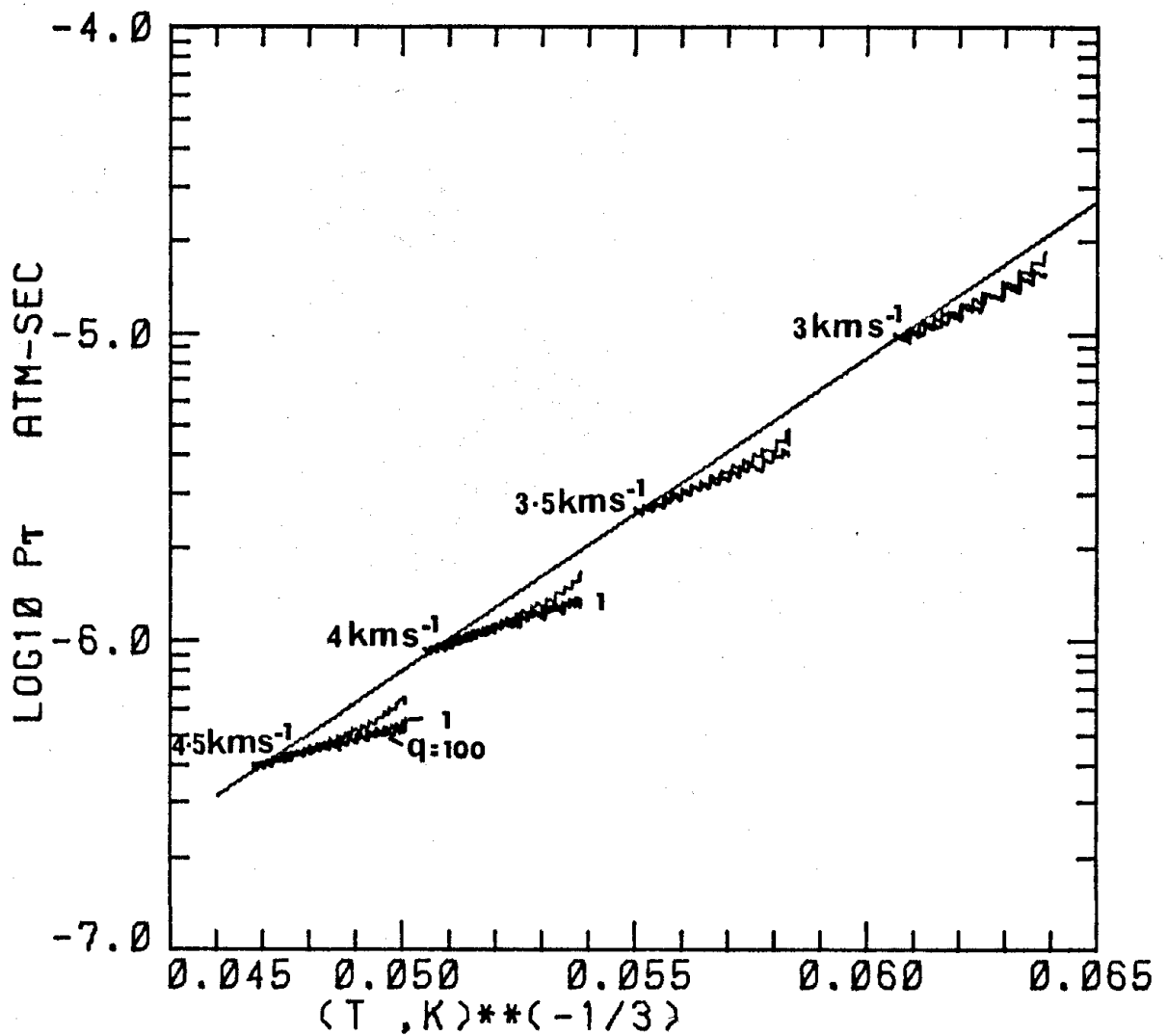


Figure 2.11: Vibrational excitation rates of N_2 . The straight line is the experimental value of Appleton (1967). Results of four shocks in 5 torr initial pressure of N_2 at 300K are plotted. The slowest rate for each shock is for V-T transitions only. The effect of V-V transitions is shown by varying q .

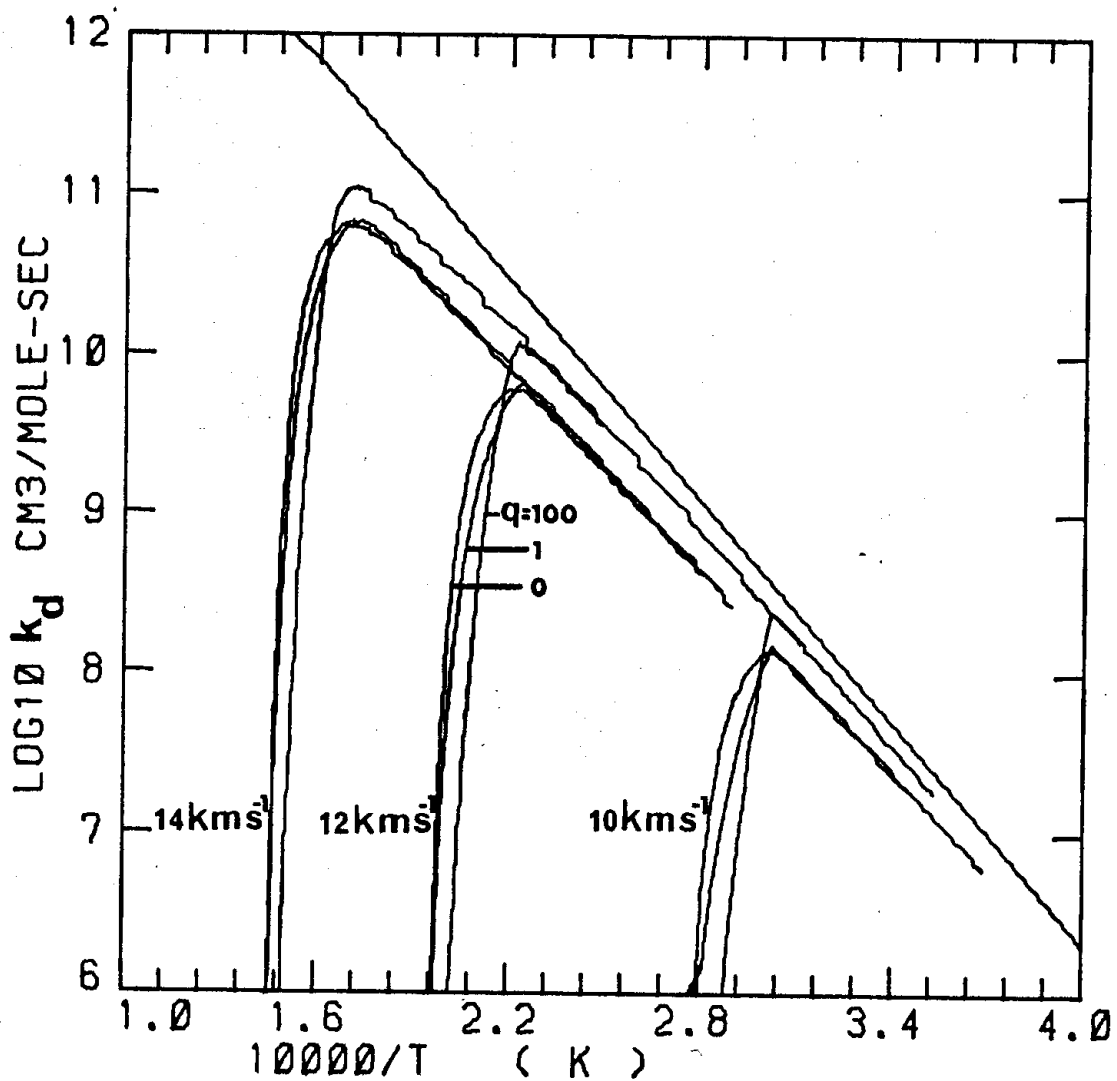


Figure 2.12: Dissociation rate coefficient k_d of $H_2 \cdot H_2$. For comparison the experimental value of k_d^2 given by Breshears and Bird (1973a) is shown. Results of three shocks in 5 torr initial pressure of H_2 at 300K are plotted. The effect of V-V transitions is shown by varying q .

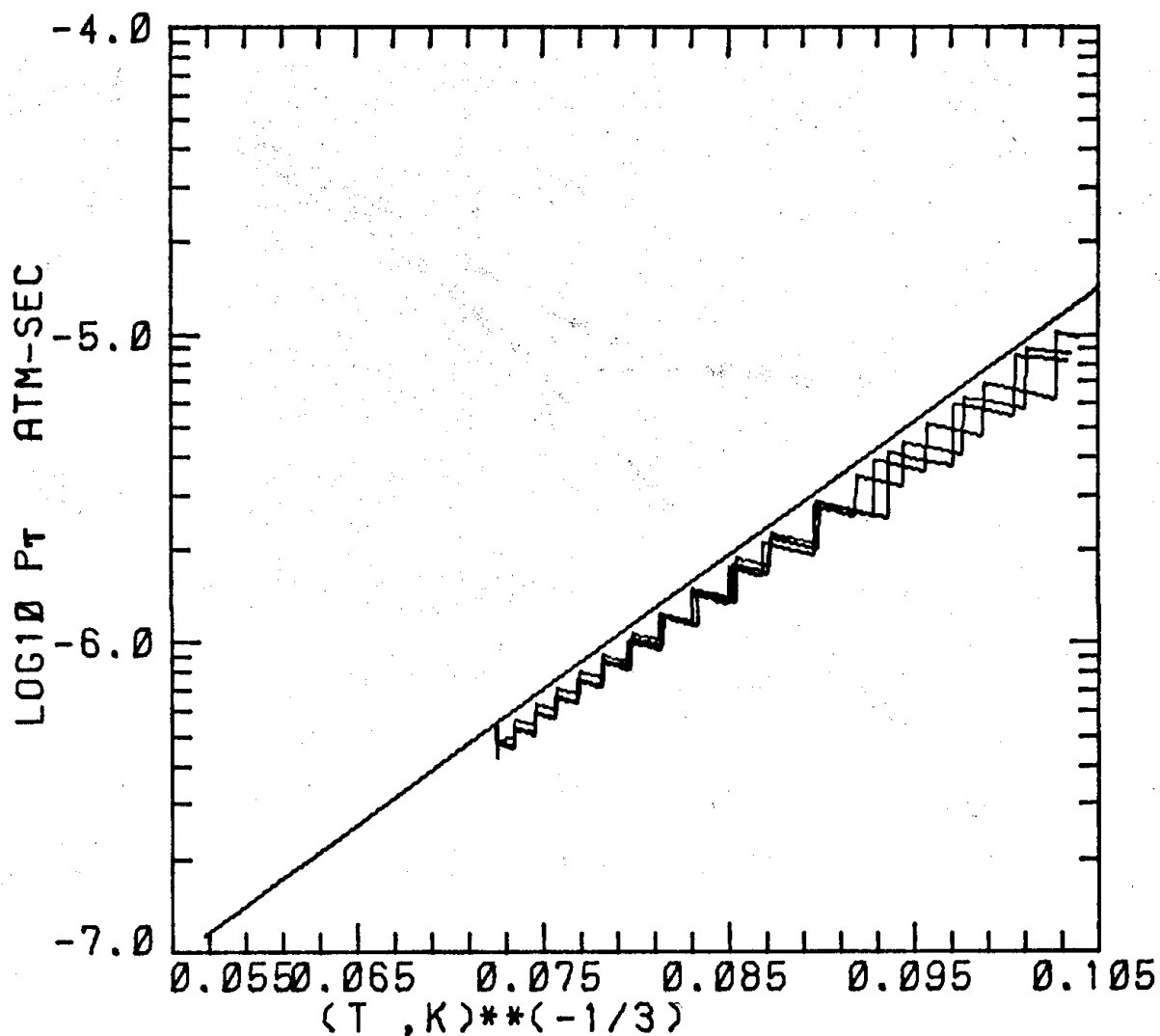


Figure 2.13: Vibrational de-excitation rates of H_2 . The straight line is the experimental value of Kiefer and Lutz (1966b). The reservoir conditions are 3000K and 50 atm. The rates for V-T transitions only, V-T and V-V transitions and V-T and V-V transitions with q equal to 10 are seen to be nearly identical.

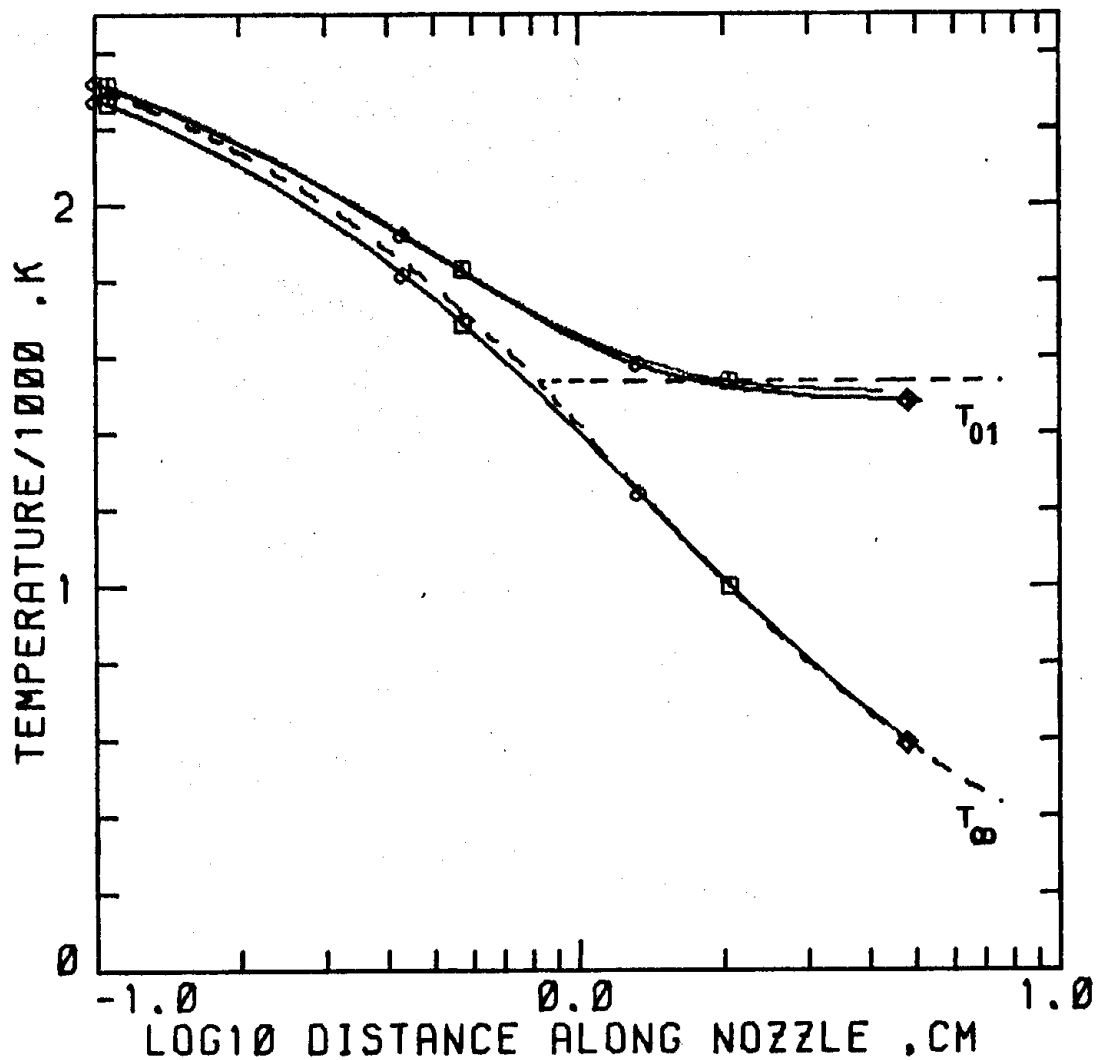


Figure 2.14: Vibrational and translational temperature along the nozzle. The reservoir conditions are 3000K and 50 atm. ---calculations using Lordi et al. (1966), —□— V-T transitions only, —◇— V-T and V-V transitions.

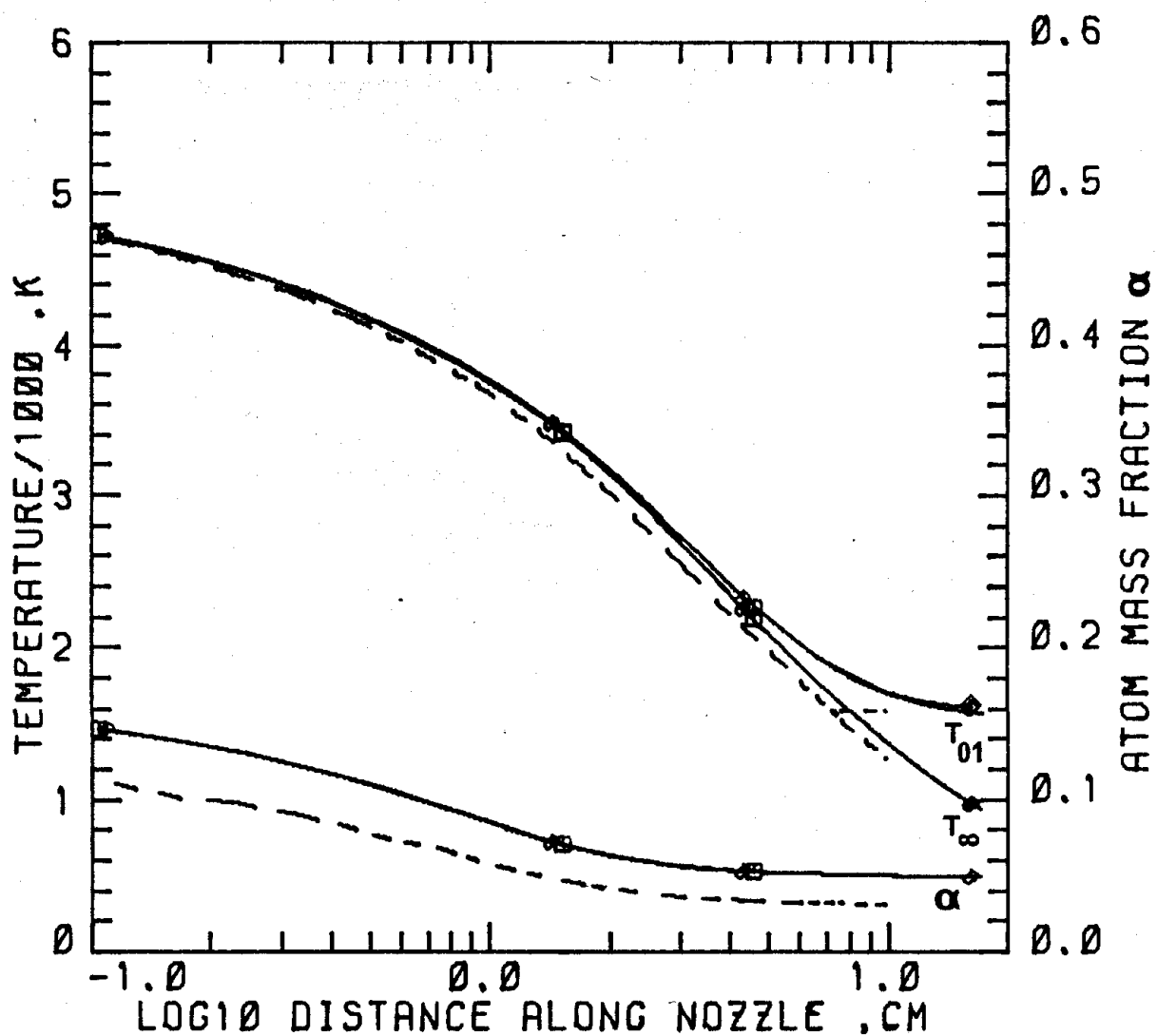


Figure 2.15a: Atom mass fraction, vibrational and translational temperature along the nozzle --- calculations using Lordi et al. (1966),
 —□— V-T transitions only, —◇— V-T and V-V transitions.
 The reservoir conditions are 5000K and 200 atm.

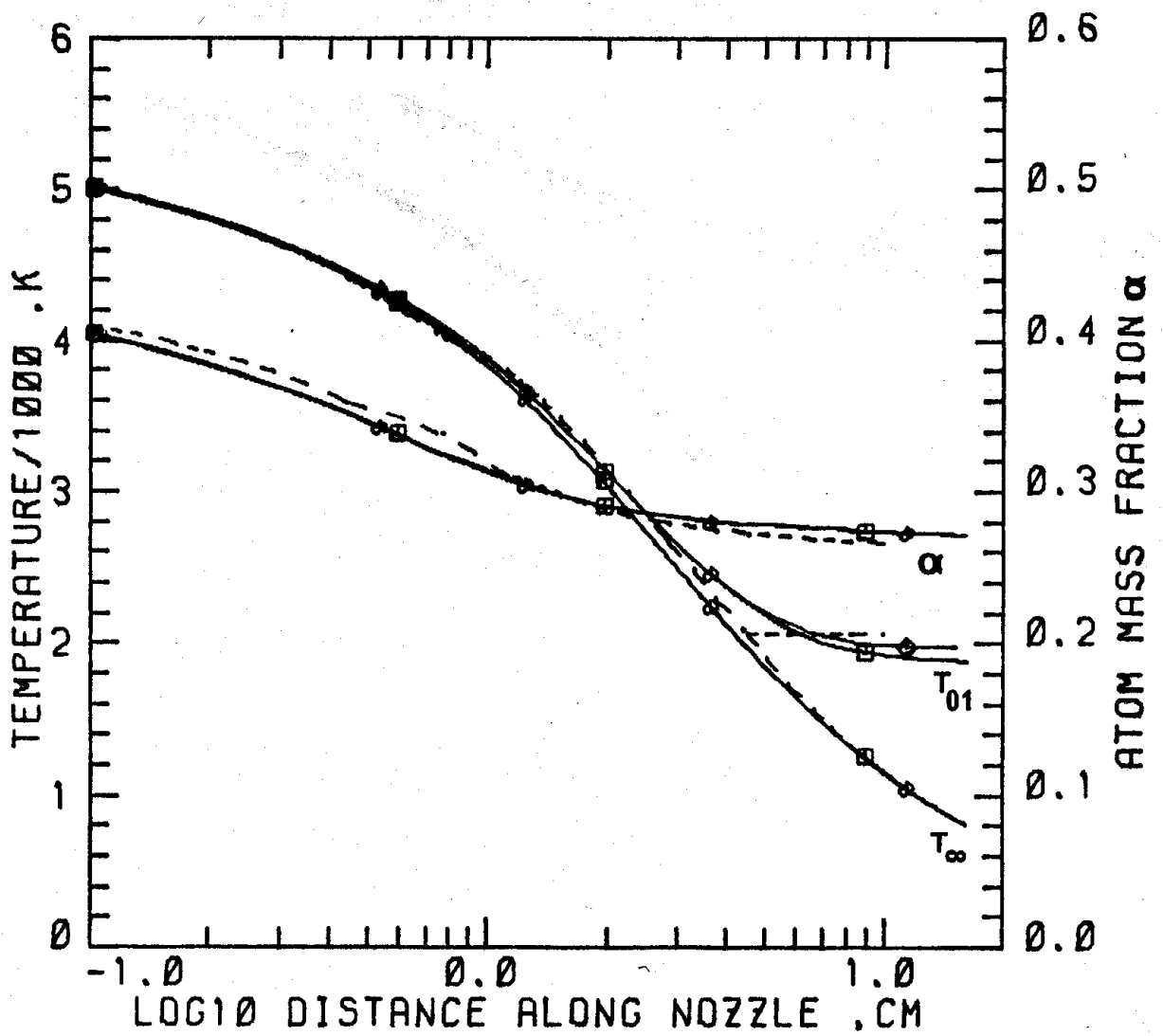


Figure 2.15b: Atom mass fraction, vibrational and translational temperature along the nozzle. The reservoir conditions are 6000K and 200 atm.

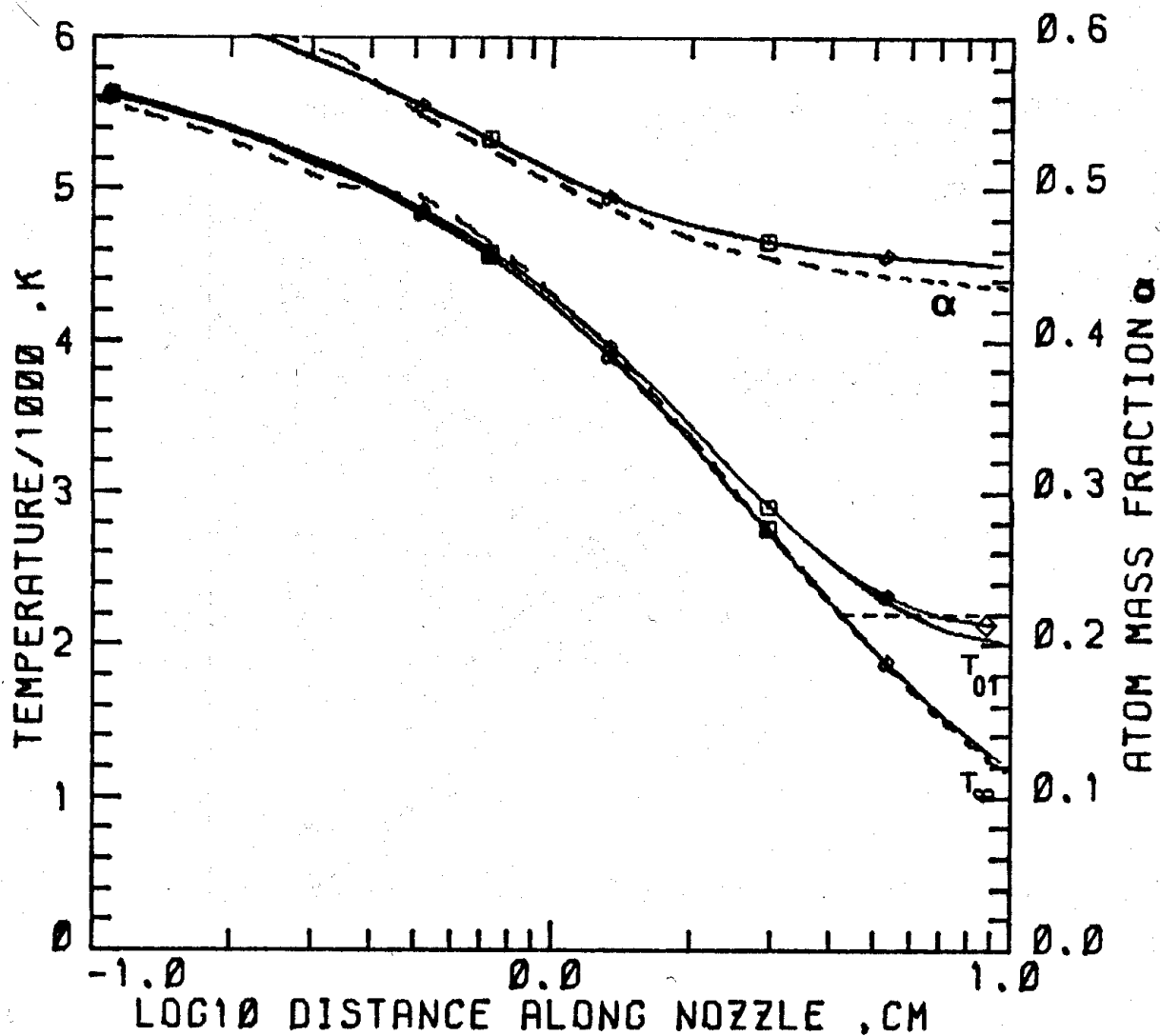


Figure 2.15c: Atom mass fraction, vibrational and translational temperature along the nozzle. The reservoir conditions are 7000K and 200 atm.

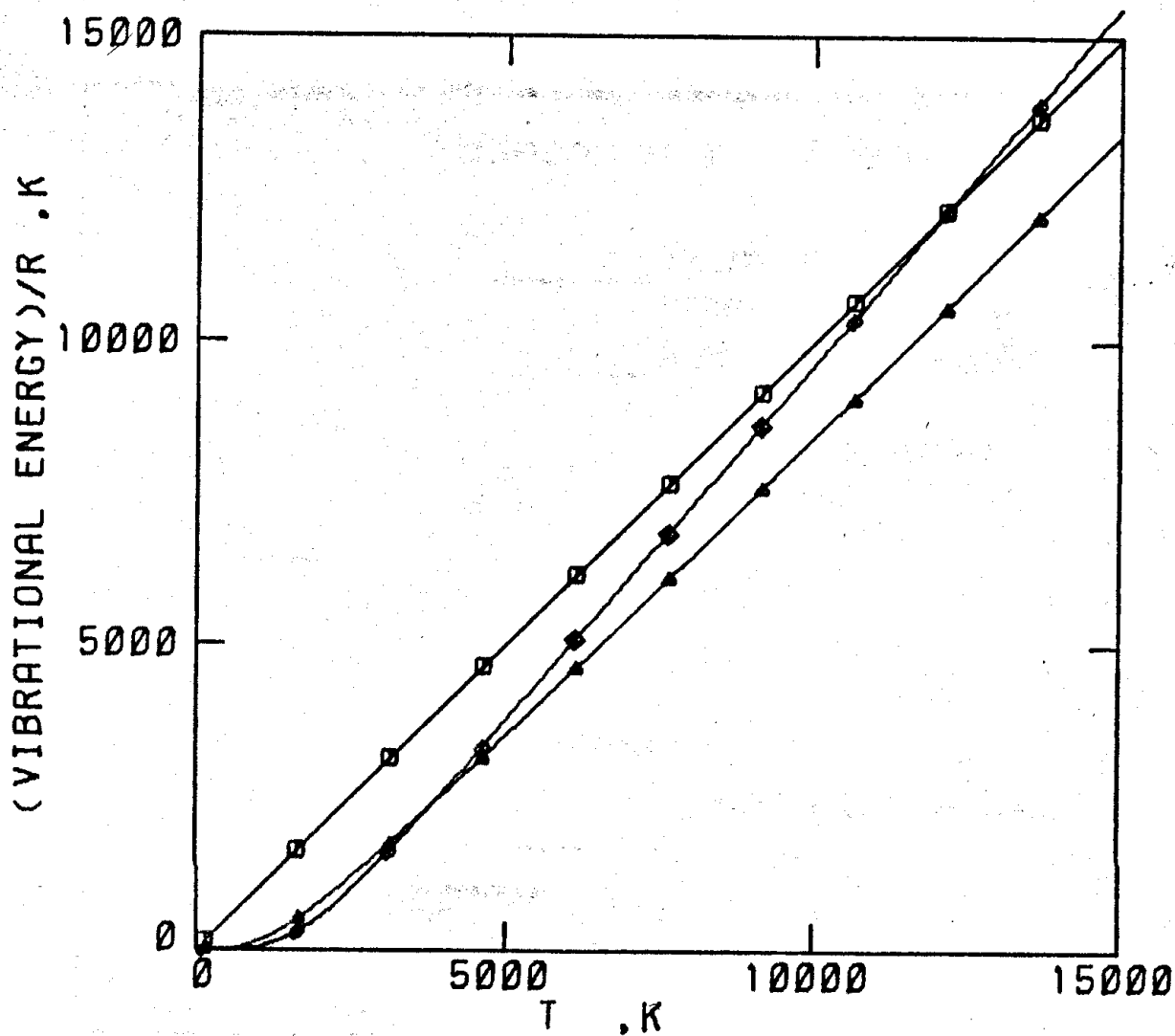


Figure 2.16: Comparison between the vibrational equilibrium energy, per mole, of the SH0 and the MO representation of N_2 for temperatures up to 15000K. \diamond MO, \blacktriangle SH0, and \square RT.

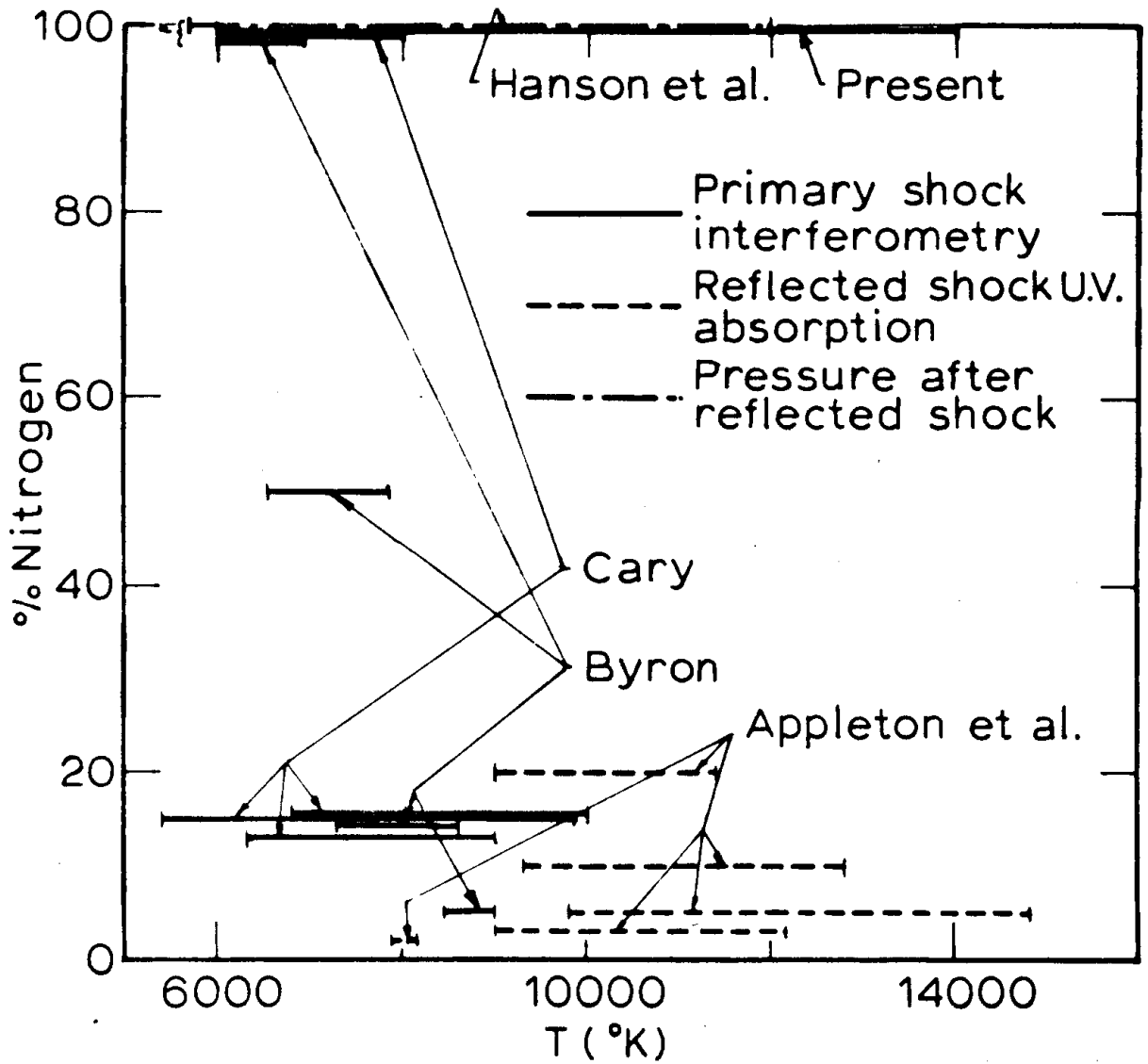


Figure 3.1: Range of temperature and inert gas dilution of different dissociation rate measurements.

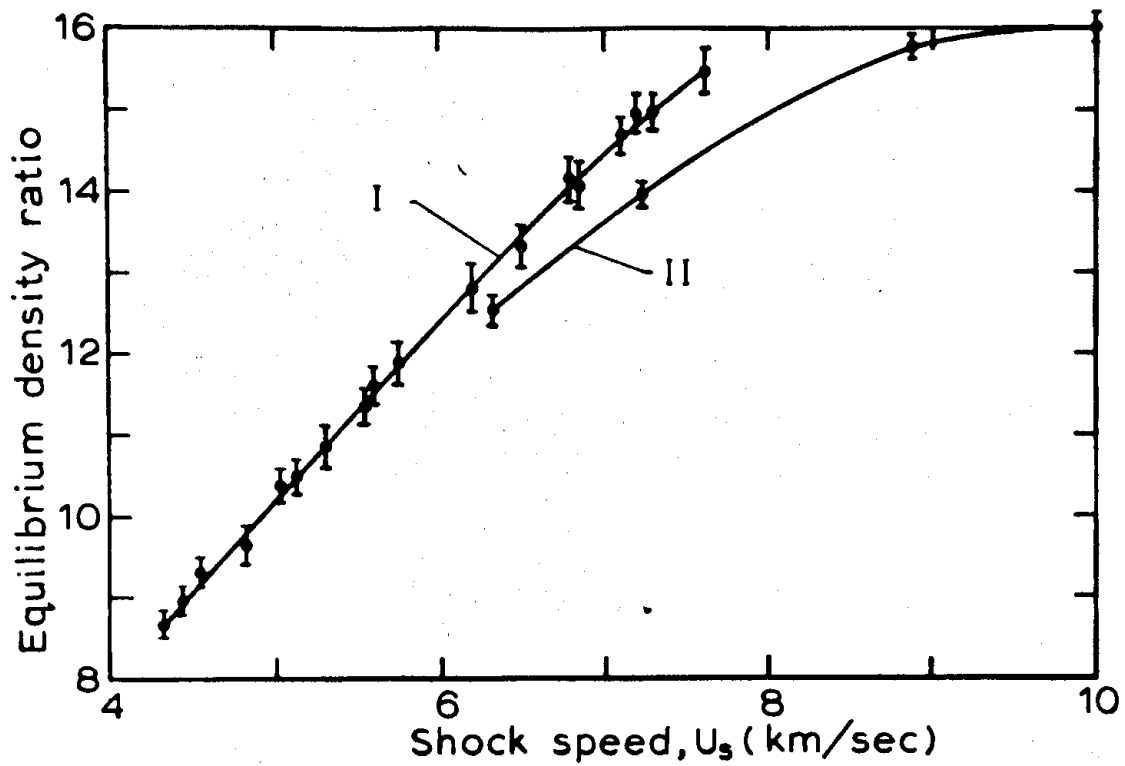
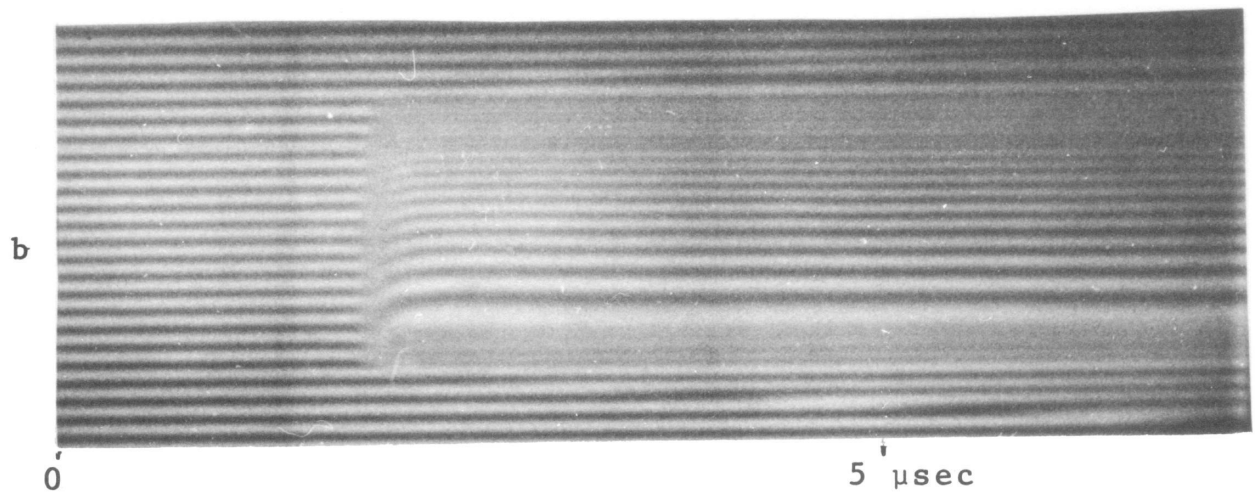
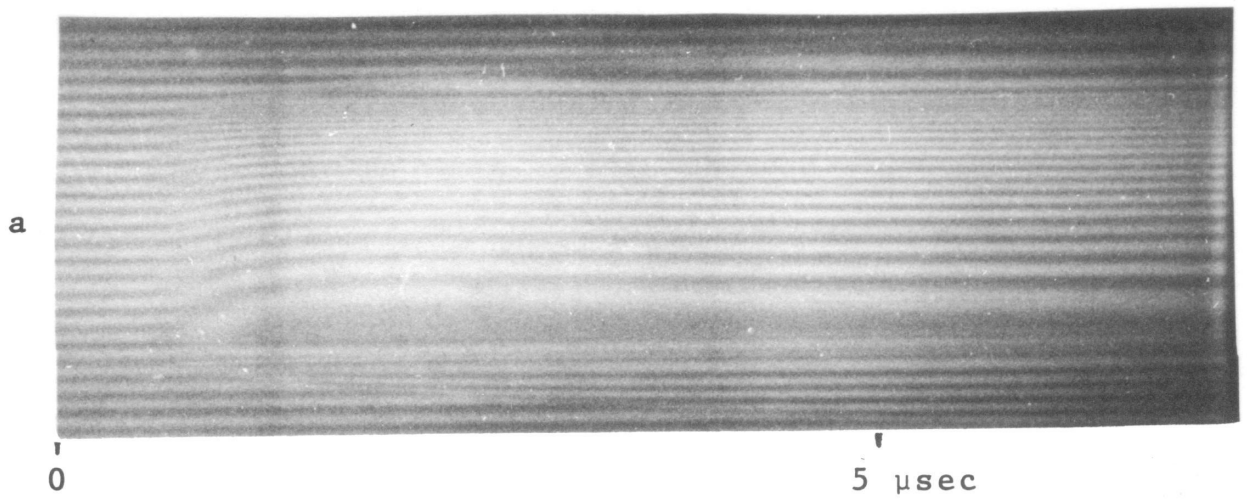


Figure 3.2: Ratio of equilibrium to initial density as a function of shock speed for two driver conditions. Curve I, 2700 lb/in², curve II, 10500 lb/in² diaphragm burst pressure. — Calculated, ● measured.



5100 Å calibration for the white light interferogram.

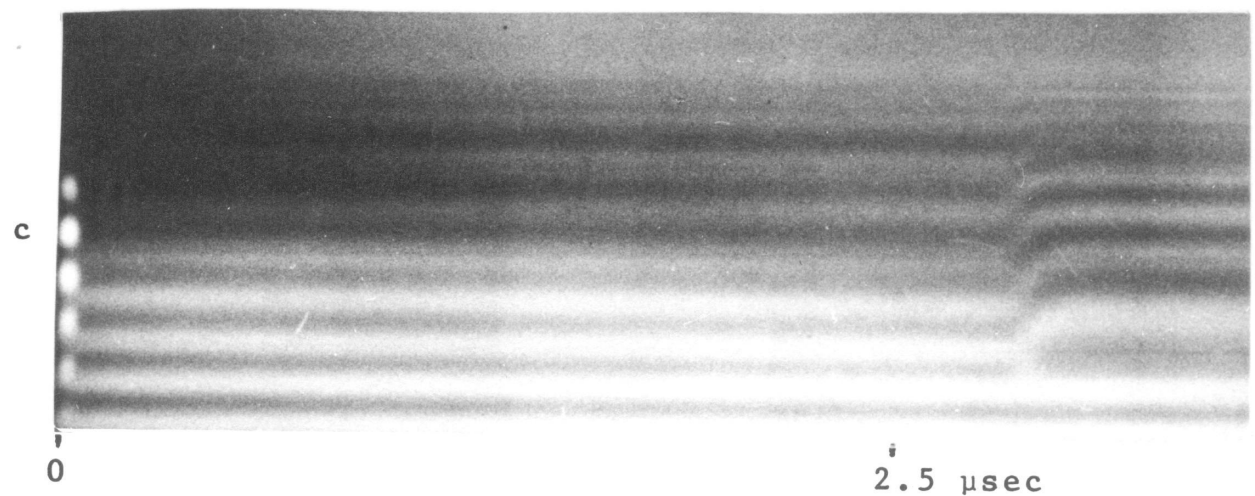


Figure 3.3: Three time-resolved interferograms of nitrogen dissociation behind a shock.

- | | |
|---|--|
| a | 4.80 km sec ⁻¹ , 31 torr, 298.2K. |
| b | 5.60 km sec ⁻¹ , 19 torr, 298.2K. |
| c | 7.31 km sec ⁻¹ , 5 torr, 300.2K. |

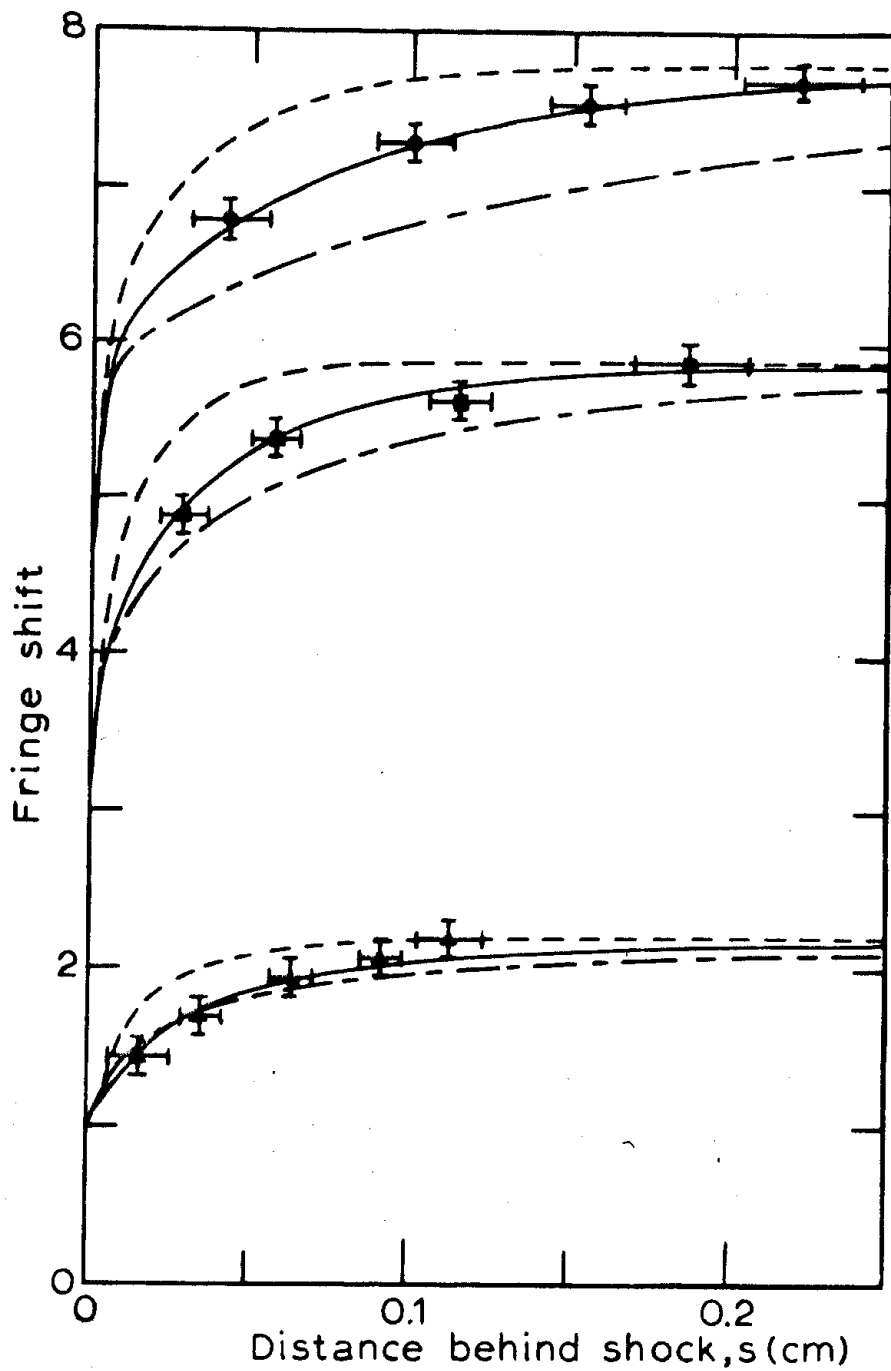


Figure 3.4: Calculated and measured fringe shift in the relaxation zone for three shock speeds: ● 4.80 km sec⁻¹, 31 torr; ■ 5.60 km sec⁻¹, 19 torr; ▲ 7.31 km sec⁻¹, 5 torr. Calculated curves according to Hanson and Baganoff (1972) ---, Appleton et al. (1968) — — — — —, equations 3.2-11 and 12

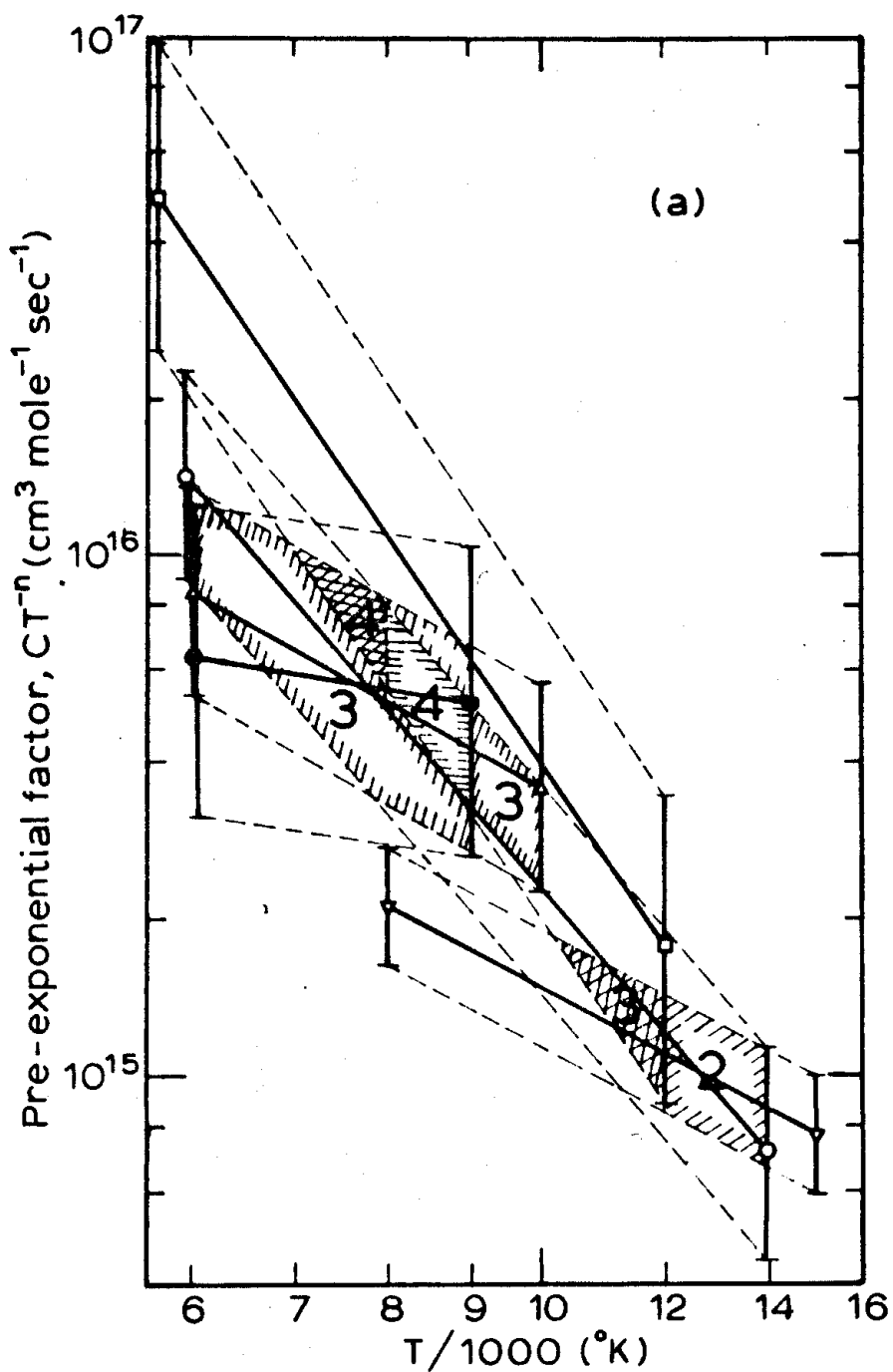


Figure 3.5a: Pre-exponential factor for the rate coefficient $k_d^{N_2}$ versus temperature; \square Hanson and Baganoff (1972), \circ present, \triangle Cary (1966), ∇ Appleton et al. (1968), \bullet Byron (1966). The error bars indicate authors' estimated uncertainties. Edge-hatching delineates regions in which 3 or more but not all, authors agree, or that if only 2 authors cover the range, they agree. Cross-hatching indicates where all authors agree. The numbers indicate how many authors agree in the region.

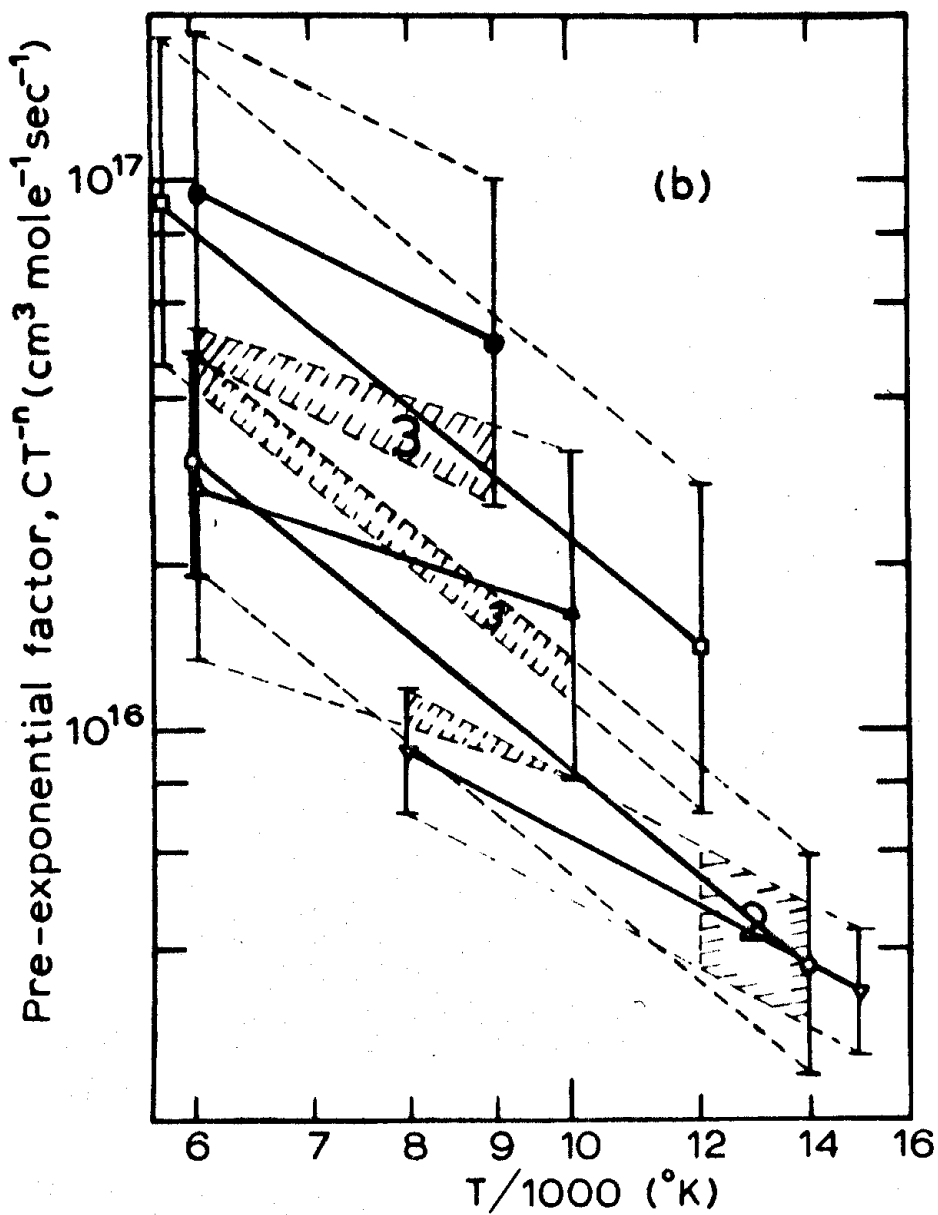


Figure 3.5b: Pre-exponential factor for the rate coefficient k_d^N versus temperature.

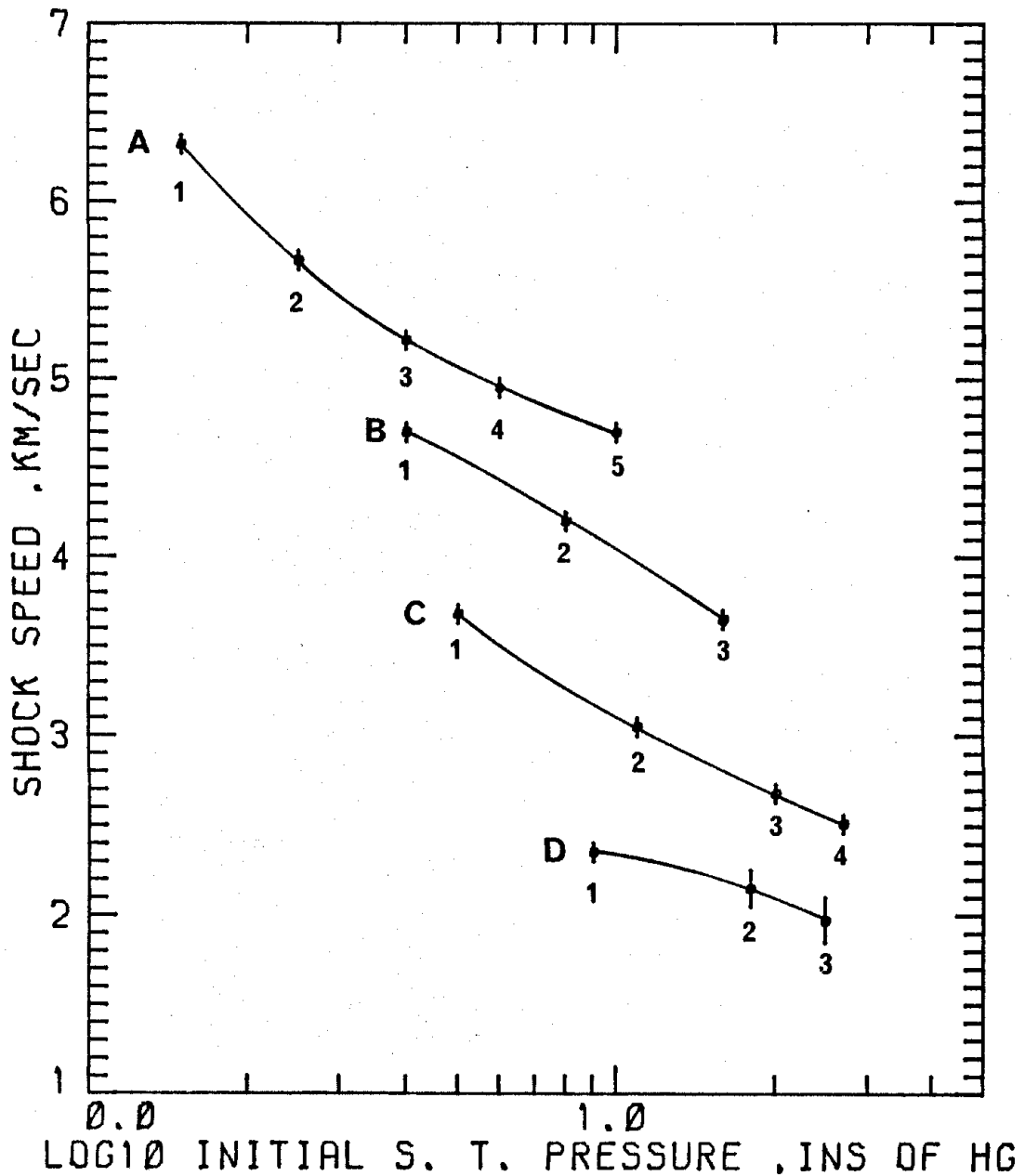
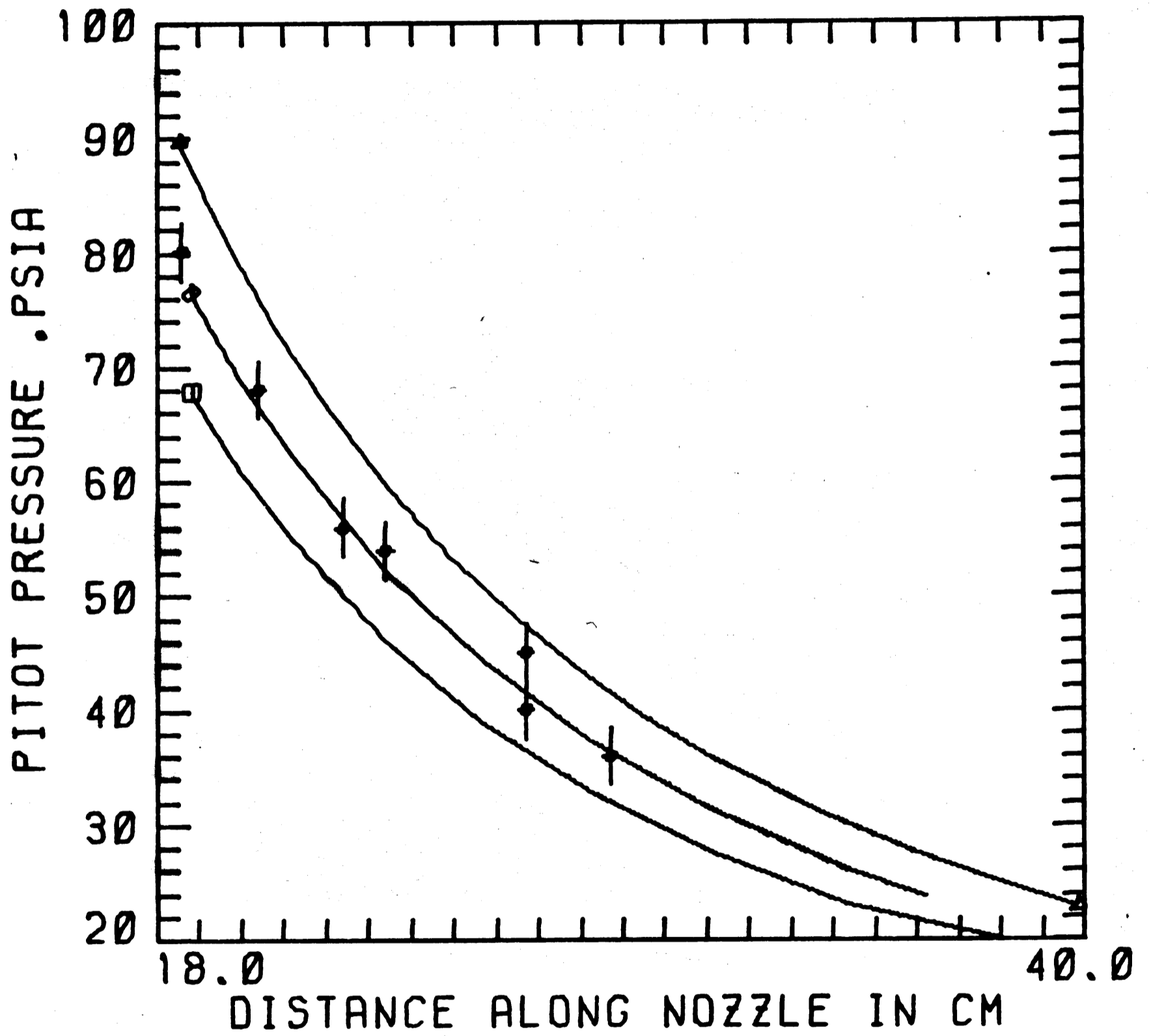
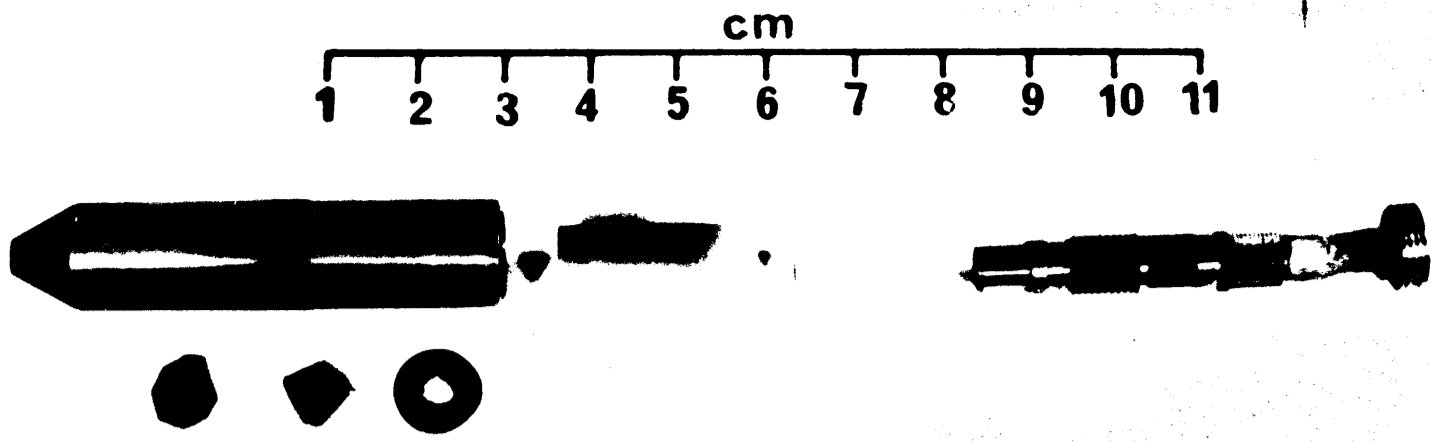


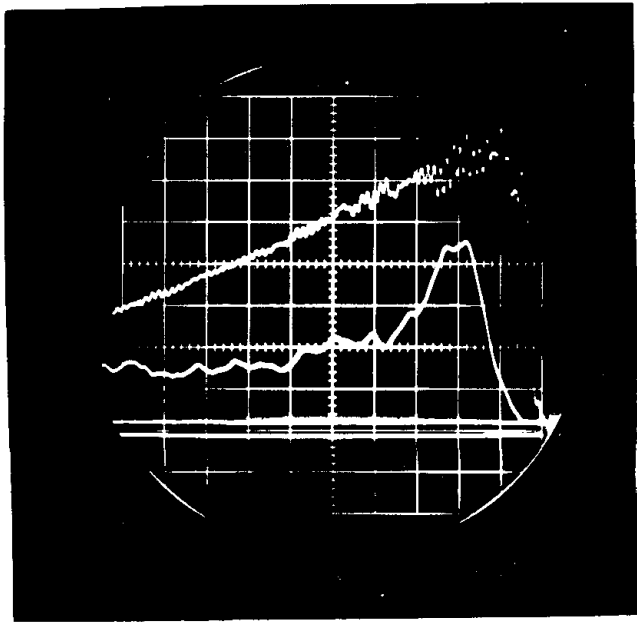
Figure 3.6: Primary shock speed in T2 versus initial shock tube pressure of nitrogen for four operating conditions

- A. He driver, $p_b = 6800$ psi, $\lambda_p = 37$
- B. He driver, $p_b = 3600$ psi, $\lambda_p = 29$
- C. He driver, $p_b = 1700$ psi, $\lambda_p = 14$
- D. Ar driver, $p_b = 6800$ psi, $\lambda_p = 37$.

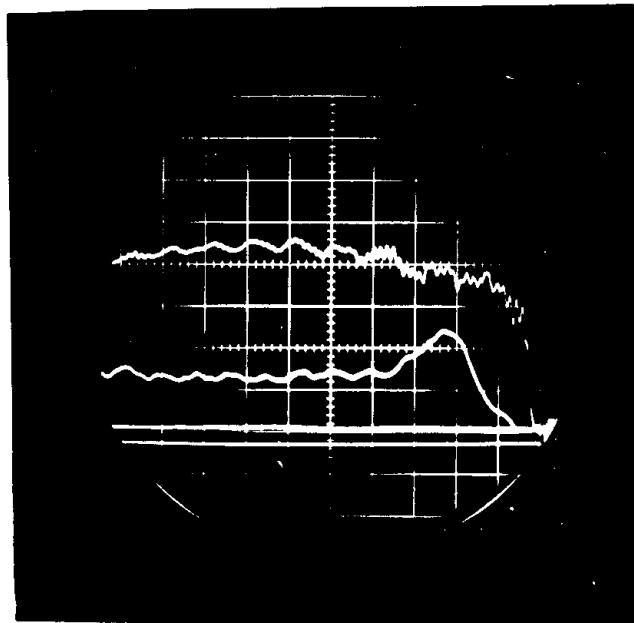


- ▲ 13 deg nozzle
- ◻ 15 deg nozzle
- ◆ 14 deg nozzle

Figure 3.7: Pitot pressure probe and results of an axial Pitot pressure survey. The nozzle stagnation conditions are $T_0 = 4400\text{K}$ and $P_0 = 227 \text{ atm}$.



Nozzle stagnation condition = B2



Nozzle stagnation condition = C1

Figure 3.8: Static pressure variation with time, measured from shock reflection in the shock tube.

Vertical scale: Stagnation pressure - 35 atm/div
 Static pressure - 0.143 psi/div

Horizontal scale: Time, increasing from right to left - 100 μ sec/div

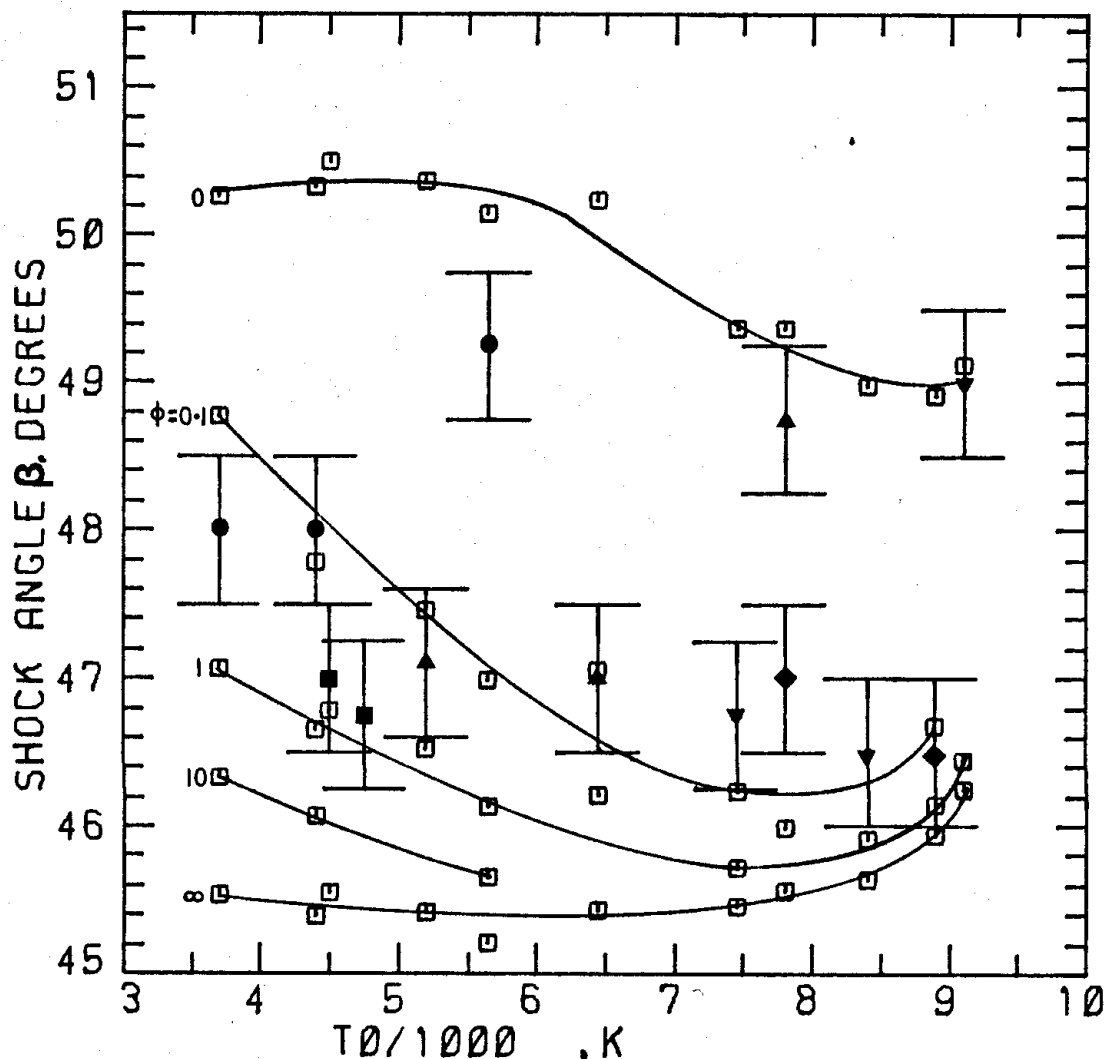


Figure 3.9: Calculated and measured shock angle on a 35° wedge for the range of stagnation temperature T_0 . — represents a free-hand fit to the calculated points \square . The time, after shock reflection, of each measured interferogram is indicated: \bullet 500 μsec , \blacktriangle 350 μsec , \blacksquare 300 μsec , \blacklozenge 225 μsec , \blacktriangledown 200 μsec . ϕ represents the ratio of vibrational de-excitation rate to excitation rate used in the calculations. The error bar of 300K for T_0 , above 4500K, is considered to be pessimistic.

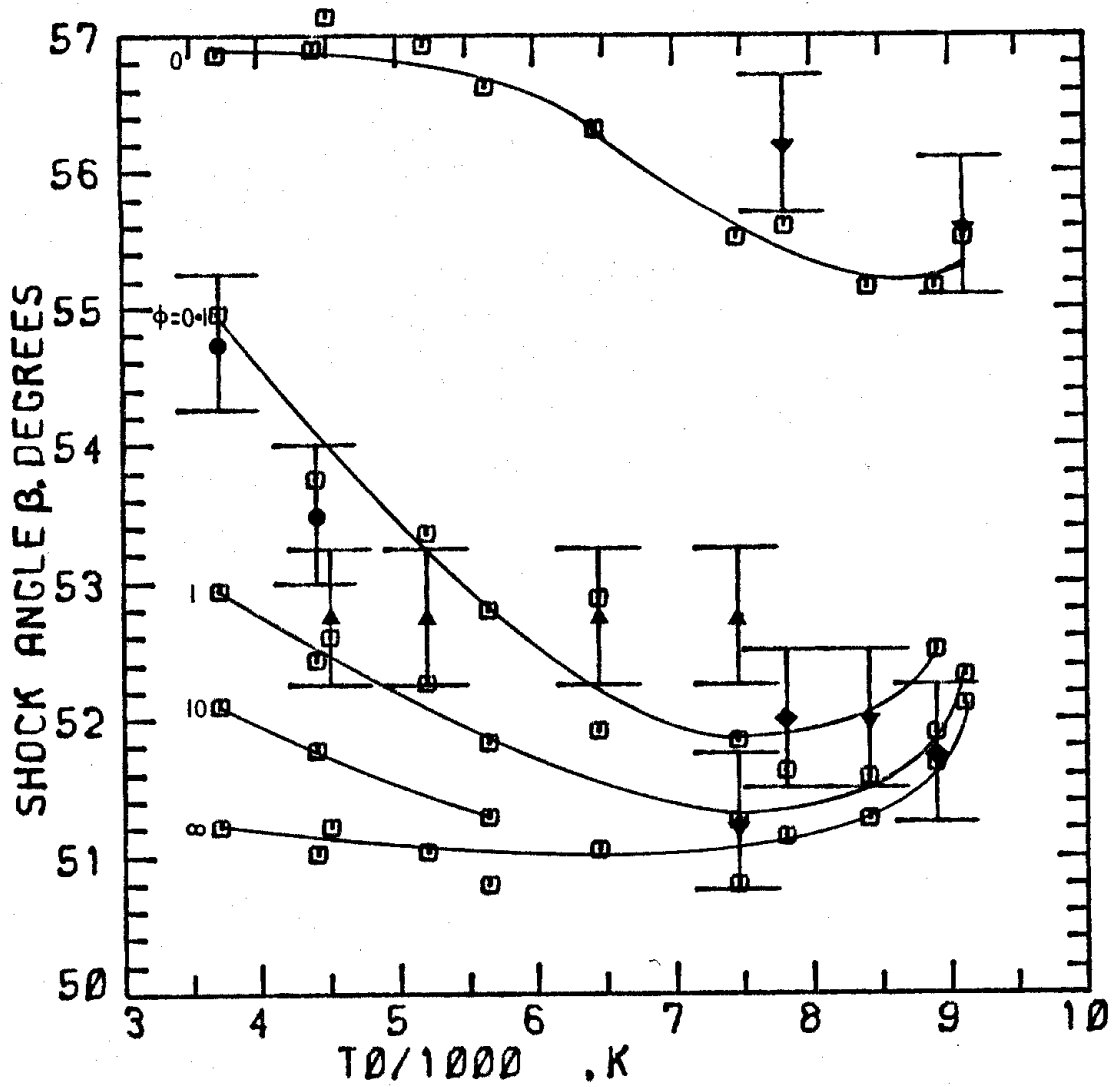


Figure 3.10: Calculated and measured shock angle on a 39° wedge for the range of stagnation temperature T_0 . The description is the same as Figure 3.9.

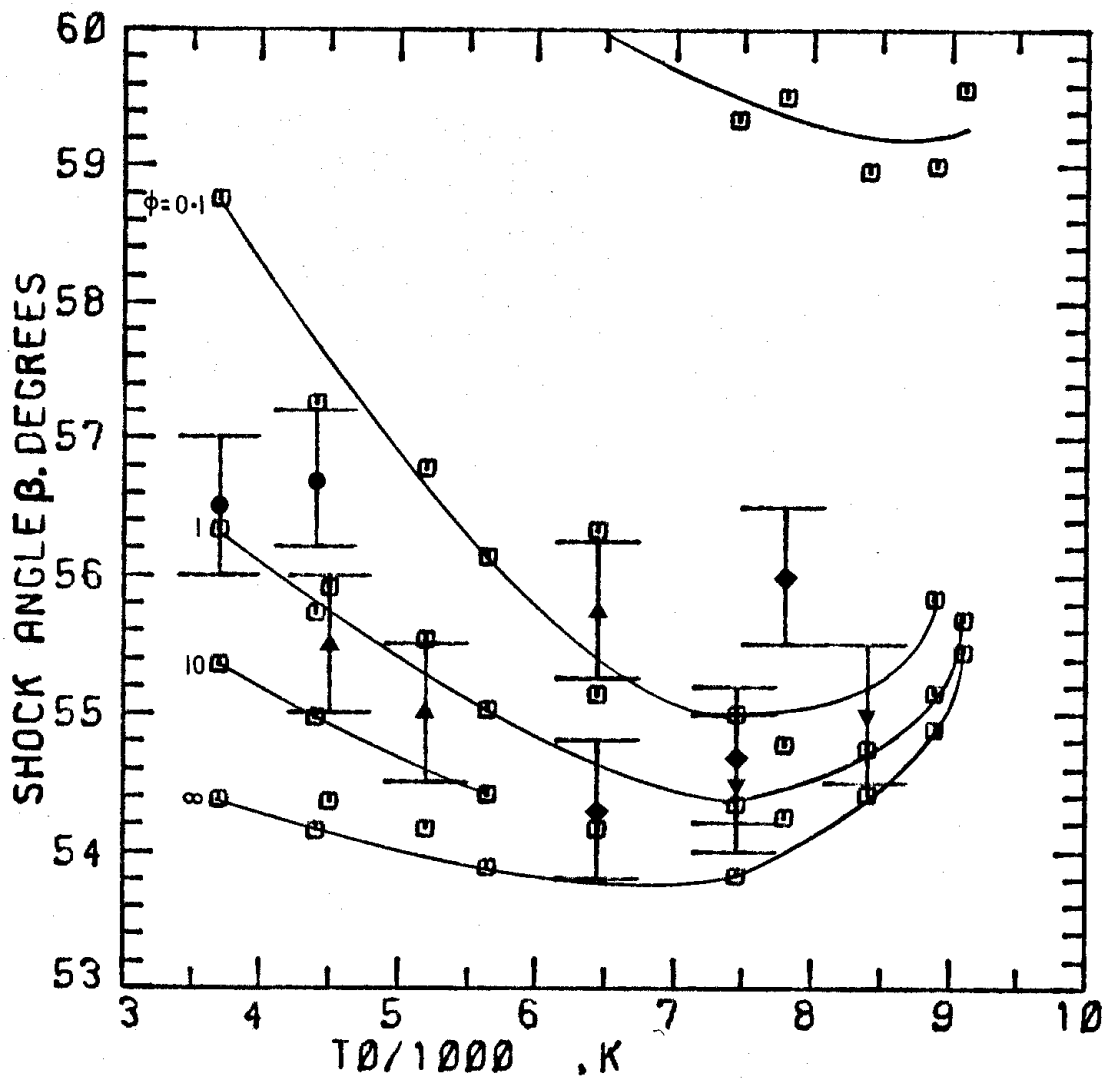


Figure 3.11: Calculated and measured shock angle on a 41° wedge for the range of stagnation temperature T_0 . The description is the same as Figure 3.9.